

(c) Affected ADs

This AD supersedes AD 2013–08–17, Amendment 39–17434 (78 FR 25380, May 1, 2013).

(d) Effective Date

This AD becomes effective June 10, 2016.

(e) Compliance

You are responsible for performing each action required by this AD within the specified compliance time unless it has already been accomplished prior to that time.

(f) Required Actions

(1) Within 110 hours time-in-service (TIS) after reaching the hours or landings threshold, whichever occurs first, listed in Table 1 to Paragraph (f)(1) of this AD or within 110 hours TIS from the effective date of this AD, whichever occurs later, and thereafter at intervals not to exceed 110 hours TIS, using a 10X or higher magnifying glass and a light, inspect the 9-degree fuselage frame on the right-hand and left-hand sides for a crack in the areas depicted in Figures 1 and 2 of Airbus Helicopters Emergency Alert Service Bulletin (EASB) No. AS365 05.00.57, Revision 2, dated April 7, 2014, or EASB No. SA366 05.39, Revision 2, dated April 7, 2014, as applicable to your model helicopter. For purposes of this AD, a landing would be counted anytime the helicopter lifts off into the air and then lands again regardless of the duration of the landing and regardless of whether the engine is shut down.

TABLE 1 TO PARAGRAPH (f)(1)

Helicopter model	Hours TIS	Landings
SA–365N	11,490	22,980
SA–365N1	10,490	20,980
AS–365N2	9,140	18,280
AS–365N3	8,740	17,480
SA–366G1	8,390	16,780

(2) If there is a crack, before further flight, repair the frame. Repairing a frame does not constitute terminating actions for the repetitive inspection requirements of this AD.

(g) Alternative Methods of Compliance (AMOCs)

(1) The Manager, Safety Management Group, FAA, may approve AMOCs for this AD. Send your proposal to: Robert Grant, Aviation Safety Engineer, Safety Management Group, FAA, 10101 Hillwood Pkwy., Fort Worth, Texas 76177; telephone (817) 222–5110; email 9-ASW-FTW-AMOC-Requests@faa.gov.

(2) For operations conducted under a 14 CFR part 119 operating certificate or under 14 CFR part 91, subpart K, we suggest that you notify your principal inspector, or lacking a principal inspector, the manager of the local flight standards district office or certificate holding district office, before operating any aircraft complying with this AD through an AMOC.

(h) Additional Information

The subject of this AD is addressed in European Aviation Safety Agency (EASA) AD No. 2014–0159, dated July 7, 2014. You may view the EASA AD on the Internet at <http://www.regulations.gov> in Docket No. FAA–2015–3741.

(i) Subject

Joint Aircraft Service Component (JASC)
Code: 5311, Fuselage Main, Frame.

(j) Material Incorporated by Reference

(1) The Director of the Federal Register approved the incorporation by reference of the service information listed in this paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) You must use this service information as applicable to do the actions required by this AD, unless the AD specifies otherwise.

(i) Airbus Helicopters Emergency Alert Service Bulletin No. 05.00.57, Revision 2, dated April 7, 2014.

(ii) Airbus Helicopters Emergency Alert Service Bulletin No. 05.39, Revision 2, dated April 7, 2014.

Note 1 to paragraph (j)(2): Airbus Helicopters Emergency Alert Service Bulletin No. 05.00.57 and Airbus Helicopters Emergency Alert Service Bulletin No. 05.39, both Revision 2, and both dated April 7, 2014, are co-published as one document along with Airbus Helicopters Emergency Alert Service Bulletin No. 05.00.25, Revision 2, dated April 7, 2014, which is not incorporated by reference in this AD.

(3) For Airbus Helicopters service information identified in this final rule, contact Airbus Helicopters, Inc., 2701 N. Forum Drive, Grand Prairie, TX 75052; telephone (972) 641–0000 or (800) 232–0323; fax (972) 641–3775; or at <http://www.airbushelicopters.com/techpub>.

(4) You may view this service information at FAA, Office of the Regional Counsel, Southwest Region, 10101 Hillwood Pkwy., Room 6N–321, Fort Worth, TX 76177. For information on the availability of this material at the FAA, call (817) 222–5110.

(5) You may view this service information that is incorporated by reference at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call (202) 741–6030, or go to: <http://www.archives.gov/federal-register/cfr/ibr-locations.html>.

Issued in Fort Worth, Texas, on April 22, 2016.

Scott A. Horn,

*Acting Manager, Rotorcraft Directorate,
Aircraft Certification Service.*

[FR Doc. 2016–10286 Filed 5–5–16; 8:45 am]

BILLING CODE 4910–13–P

DEPARTMENT OF TRANSPORTATION**Federal Aviation Administration****14 CFR Part 39**

[Docket No. FAA–2015–8427; Directorate Identifier 2014–NM–212–AD; Amendment 39–18508; AD 2016–09–10]

RIN 2120–AA64

Airworthiness Directives; Airbus Airplanes

AGENCY: Federal Aviation Administration (FAA), Department of Transportation (DOT).

ACTION: Final rule.

SUMMARY: We are superseding Airworthiness Directive (AD) 2007–10–10 R1 for all Airbus Model A300 B4–600, B4–600R, and F4–600R series airplanes, and Model A300 C4–605R Variant F airplanes (collectively called Model A300–600 series airplanes). AD 2007–10–10 R1 required revising the Airworthiness Limitations Section (ALS) of the Instructions for Continued Airworthiness to incorporate new limitations for fuel tank systems. This new AD requires revising the maintenance program or inspection program to incorporate revised fuel maintenance and inspection tasks. This AD was prompted by issuance of more restrictive maintenance requirements and/or airworthiness limitations by the manufacturer. We are issuing this AD to prevent the potential of ignition sources inside fuel tanks, which, in combination with flammable fuel vapors caused by latent failures, alterations, repairs, or maintenance actions, could result in fuel tank explosions and consequent loss of the airplane.

DATES: This AD becomes effective June 10, 2016.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in this AD as of June 10, 2016.

The Director of the Federal Register approved the incorporation by reference of a certain other publication listed in this AD as of December 28, 2009 (74 FR 65398, December 10, 2009).

The Director of the Federal Register approved the incorporation by reference of certain other publications listed in this AD as of June 27, 2007 (72 FR 28827, May 23, 2007).

ADDRESSES: For service information identified in this final rule, contact Airbus SAS, Airworthiness Office—EAW, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France; telephone +33 5 61 93 36 96; fax +33 5 61 93 44 51; email account.airworth-eas@

[airbus.com](http://www.airbus.com); Internet <http://www.airbus.com>. You may view this referenced service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, WA. For information on the availability of this material at the FAA, call 425-227-1221. It is also available on the Internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA-2015-8427.

Examining the AD Docket

You may examine the AD docket on the Internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA-2015-8427; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this AD, the regulatory evaluation, any comments received, and other information. The address for the Docket Office (telephone 800-647-5527) is Docket Management Facility, U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE., Washington, DC 20590.

FOR FURTHER INFORMATION CONTACT: Dan Rodina, Aerospace Engineer, International Branch, ANM-116, Transport Airplane Directorate, FAA, 1601 Lind Avenue SW., Renton, WA 98057-3356; telephone 425-227-2125; fax 425-227-1149.

SUPPLEMENTARY INFORMATION:

Discussion

We issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 to supersede AD 2007-10-10 R1, Amendment 39-16134 (74 FR 65398, December 10, 2009) (“AD 2007-10-10 R1”). AD 2007-10-10 R1 applied to all Airbus Model A300 B4-600, B4-600R, and F4-600R series airplanes, and Model A300 C4-605R Variant F airplanes (collectively called Model A300-600 series airplanes). The NPRM published in the **Federal Register** on January 13, 2016 (81 FR 1573) (“the NPRM”).

The European Aviation Safety Agency (EASA), which is the Technical Agent for the Member States of the European Union, has issued EASA Airworthiness Directive 2014-0194, dated October 15, 2014 (referred to after this as the Mandatory Continuing Airworthiness Information, or “the MCAI”), to correct an unsafe condition on all Airbus Model A300 B4-600, B4-600R, and F4-600R series airplanes, and Model A300 C4-605R Variant F airplanes (collectively called Model A300-600 series airplanes). The MCAI states:

Prompted by an accident * * *, the Federal Aviation Administration (FAA) published Special Federal Aviation Regulation (SFAR) 88, [[http://rgl.faa.gov/Regulatory and Guidance Library/rgFAR.nsf/0/EEFB3F94451DC06286256C93004F5E07?OpenDocument&Highlight=sfar](http://rgl.faa.gov/Regulatory%20and%20Guidance%20Library/rgFAR.nsf/0/EEFB3F94451DC06286256C93004F5E07?OpenDocument&Highlight=sfar) 88], and the Joint Aviation Authorities (JAA) published Interim Policy INT/POL/25/12. In response to these regulations, Airbus conducted a design review to develop Fuel Airworthiness Limitations (FAL) for Airbus on A300-600 and A300-600ST aeroplanes.

The FAL were specified in Airbus A300-600 FAL document ref. 95A.1929/05 at issue 02 and in the A300-600 [Airworthiness Limitation Section] ALS variation to FAL document issue 02 ref. 0CVLG110007/COS issue 01, for A300-600 and A300-600ST aeroplanes.

EASA issued [EASA] AD 2006-0201 to require compliance with the FAL documents (comprising maintenance/inspection tasks and Critical Design Configuration Control Limitations (CDCCL)).

EASA AD 2006-0201 was superseded by EASA AD 2007-0095 (later revised) [which corresponds to FAA AD 2007-10-10 R1, Amendment 39-16134 (74 FR 65398, December 10, 2009)], which retained the original requirements and corrected and updated the compliance paragraphs concerning task ref. 28-18-00-03-1 and CDCCLs.

Since EASA AD 2007-0095R1 was published, Airbus issued A300-600 ALS Part 5, prompted by EASA policy statement (EASA D2005/CPRO) which requests design approval holders to integrate Fuel Tank Safety items into an ALS document. The A300-600 ALS Part 5 is approved by EASA.

Failure to comply with the items as identified in Airbus A300-600 ALS Part 5 could result in a fuel tank explosion and consequent loss of the aeroplane.

For the reasons described above, this [EASA] AD retains the requirements of EASA AD 2007-0095R1, which is superseded, and requires implementation of the new and more restrictive maintenance instructions and/or airworthiness limitations as specified in Airbus A300-600 ALS Part 5.

The unsafe condition is the potential of ignition sources inside fuel tanks, which, in combination with flammable fuel vapors caused by latent failures, alterations, repairs, or maintenance actions, could result in fuel tank explosions and consequent loss of the airplane. You may examine the MCAI in the AD docket on the Internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA-2015-8427.

Comments

We gave the public the opportunity to participate in developing this AD. We received no comments on the NPRM or on the determination of the cost to the public.

Explanation of Changes Made in This AD

We have corrected certain paragraph references in paragraph (g) and Note 1 to paragraph (g) of this AD.

We have revised certain document citations in paragraph (h) of this AD. This change is necessary to meet the Office of the Federal Register's requirements for documents that are incorporated by reference in this AD.

We have also determined that the compliance time for revising the ALS to incorporate certain CDCCLs specified in paragraph (i) of this AD should be within 12 months after June 27, 2007 (the effective date of AD 2007-10-10, Amendment 39-15051 (72 FR 28827, May 23, 2007) (“AD 2007-10-10”). The compliance time stated in the proposed AD was “Within 12 months after the effective date of this AD.” We have revised paragraph (i) of this AD accordingly. The proposed AD clearly indicated that we intended to restate the requirements of that paragraph from the previous AD, with no changes.

Conclusion

We reviewed the relevant data and determined that air safety and the public interest require adopting this AD with the changes described previously, except for minor editorial changes. We have determined that these minor changes:

- Are consistent with the intent that was proposed in the NPRM for correcting the unsafe condition; and
- Do not add any additional burden upon the public than was already proposed in the NPRM.

Related Service Information Under 1 CFR Part 51

Airbus has issued A300-600 Airworthiness Limitations Section Part 5—Fuel Airworthiness Limitations, Revision 00, dated May 27, 2014. The airworthiness limitations introduce mandatory instructions and more restrictive maintenance requirements. This service information is reasonably available because the interested parties have access to it through their normal course of business or by the means identified in the **ADDRESSES** section.

Costs of Compliance

We estimate that this AD affects 122 airplanes of U.S. registry.

The actions required by AD 2007-10-10 R1, and retained in this AD take about 2 work-hours per product, at an average labor rate of \$85 per work-hour. Based on these figures, the estimated cost of the actions that were required by AD 2007-10-10 R1 is \$170 per product.

We also estimate that it takes about 1 work-hour per product to comply with the basic requirements of this AD. The average labor rate is \$85 per work-hour. Required parts will cost \$0 per product. Based on these figures, we estimate the cost of this AD on U.S. operators to be \$10,370, or \$85 per product.

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. "Subtitle VII: Aviation Programs," describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in "Subtitle VII, Part A, Subpart III, Section 44701: General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

We determined that this AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that this AD:

1. Is not a "significant regulatory action" under Executive Order 12866;
2. Is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979);
3. Will not affect intrastate aviation in Alaska; and
4. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

- 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

- 2. The FAA amends § 39.13 by removing Airworthiness Directive (AD) 2007–10–10 R1, Amendment 39–16134 (74 FR 65398, December 10, 2009), and adding the following new AD:

2016–09–10 Airbus: Amendment 39–18508. Docket No. FAA–2015–8427; Directorate Identifier 2014–NM–212–AD.

(a) Effective Date

This AD becomes effective June 10, 2016.

(b) Affected ADs

This AD replaces AD 2007–10–10 R1, Amendment 39–16134 (74 FR 65398, December 10, 2009) ("AD 2007–10–10 R1").

(c) Applicability

This AD applies to Airbus Model A300 B4–600, B4–600R, and F4–600R series airplanes, and Model A300 C4–605R Variant F airplanes (collectively called Model A300–600 series airplanes), certificated in any category, all manufacturer serial numbers.

(d) Subject

Air Transport Association (ATA) of America Code 05, Time Limits/Maintenance Checks.

(e) Reason

This AD was prompted by Airbus issuing more restrictive instructions and/or fuel airworthiness limitations. We are issuing this AD to prevent the potential of ignition sources inside fuel tanks, which, in combination with flammable fuel vapors caused by latent failures, alterations, repairs, or maintenance actions, could result in fuel tank explosions and consequent loss of the airplane.

(f) Compliance

Comply with this AD within the compliance times specified, unless already done.

(g) Retained Revision of the Airworthiness Limitations Section (ALS) To Incorporate Fuel Maintenance and Inspection Tasks, With Corrected Paragraph References

This paragraph restates the requirements of paragraph (f) of AD 2007–10–10 R1, with corrected paragraph references. Within 3 months after June 27, 2007 (the effective date of AD 2007–10–10, Amendment 39–15051 (72 FR 28827, May 23, 2007) ("AD 2007–10–10")), revise the ALS of the Instructions for Continued Airworthiness to incorporate Airbus A300–600 ALS Part 5—Fuel Airworthiness Limitations, dated May 31, 2006, as defined in Section 1, "Maintenance/Inspection Tasks," of Airbus A300–600 Fuel Airworthiness Limitations, Document 95A.1929/05, Issue 1, dated December 19, 2005; or Airbus A300–600 Fuel Airworthiness Limitations, Document

95A.1929/05, Issue 2, dated May 16, 2007. For all tasks identified in Section 1 of these documents, the initial compliance times start from the later of the times specified in paragraphs (g)(1) and (g)(2) of this AD, and the repetitive inspections must be accomplished thereafter at the intervals specified in Section 1 of these documents, except as provided by paragraph (h) of this AD.

(1) June 27, 2007 (the effective date of AD 2007–10–10).

(2) The date of issuance of the original French standard airworthiness certificate or the date of issuance of the original French export certificate of airworthiness.

Note 1 to paragraph (g) of this AD: Airbus Operator Information Telex (OIT) SE 999.0076/06, dated June 20, 2006, identifies the applicable sections of the Airbus A300–600 Airplane Maintenance Manual for accomplishing the tasks specified in Section 1 of Document 95A.1929/05.

(h) Retained Revision of Initial Compliance Time for Task 28–18–00–03–1, With Revised Document Citations

This paragraph restates the requirements of paragraph (g) of AD 2007–10–10 R1, with revised document citations. For Task 28–18–00–03–1, "Operational check of low-level/underfull/calibration sensors," identified in Section 1, "Maintenance/Inspection Tasks" of Airbus A300–600 Fuel Airworthiness Limitations, Document 95A.1929/05, Issue 1, dated December 19, 2005; or Airbus A300–600 Fuel Airworthiness Limitations, Document 95A.1929/05, Issue 2, dated May 16, 2007: The initial compliance time is the later of the times specified in paragraphs (h)(1) and (h)(2) of this AD. Thereafter, Task 28–18–00–03–1 must be accomplished at the repetitive interval specified in Section 1 of these documents.

(1) Prior to the accumulation of 40,000 total flight hours.

(2) Within 72 months or 20,000 flight hours after June 27, 2007 (the effective date of AD 2007–10–10), whichever occurs first.

(i) Retained Revision of the ALS To Incorporate CDCCLs, With Revised Compliance Time

This paragraph restates the requirements of paragraph (h) of AD 2007–10–10 R1, with a revised compliance time. Within 12 months after June 27, 2007 (the effective date of AD 2007–10–10), revise the ALS of the Instructions for Continued Airworthiness to incorporate Airbus A300–600 ALS Part 5—Fuel Airworthiness Limitations, dated May 31, 2006, as defined in Section 2, "Critical Design Configuration Control Limitations," of Airbus A300–600 Fuel Airworthiness Limitations, Document 95A.1929/05, Issue 1, dated December 19, 2005; or Airbus A300–600 Fuel Airworthiness Limitations, Document 95A.1929/05, Issue 2, dated May 16, 2007.

(j) New Requirement of This AD: Revise the Maintenance or Inspection Program

Within 3 months after the effective date of this AD, revise the maintenance or inspection program, as applicable, by incorporating the airworthiness limitations as specified in

Airbus A300–600 Airworthiness Limitations Section Part 5—Fuel Airworthiness Limitations, Revision 00, dated May 27, 2014. The initial compliance times for the actions specified in Airbus A300–600 Airworthiness Limitations Section Part 5—Fuel Airworthiness Limitations, Revision 00, dated May 27, 2014, are at the later of the times specified in Airbus A300–600 Airworthiness Limitations Section Part 5—Fuel Airworthiness Limitations, Revision 00, dated May 27, 2014, or within 3 months after the effective date of this AD, whichever occurs later. Accomplishing the revision required by this paragraph terminates the actions required by paragraphs (g) through (i) of this AD.

(k) New Requirement of This AD: No Alternative Actions, Intervals, and/or CDCCLs

After the maintenance or inspection program has been revised as required by paragraph (j) of this AD, no alternative actions (e.g., inspections), intervals, and/or CDCCLs may be used unless the actions, intervals, and/or CDCCLs are approved as an alternative method of compliance (AMOC) in accordance with the procedures specified in paragraph (l)(1) of this AD.

(l) Other FAA AD Provisions

The following provisions also apply to this AD:

(1) *Alternative Methods of Compliance (AMOCs)*: The Manager, International Branch, ANM–114, Transport Airplane Directorate, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the International Branch, send it to ATTN: Dan Rodina, Aerospace Engineer, International Branch, ANM–116, Transport Airplane Directorate, FAA, 1601 Lind Avenue SW., Renton, WA 98057–3356; telephone 425–227–2125; fax 425–227–1149. Information may be emailed to: 9-ANM-116-AMOC-REQUESTS@faa.gov. Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office. The AMOC approval letter must specifically reference this AD.

(2) *Contacting the Manufacturer*: As of the effective date of this AD, for any requirement in this AD to obtain corrective actions from a manufacturer, the action must be accomplished using a method approved by the Manager, International Branch, ANM–116, Transport Airplane Directorate, FAA; or the European Aviation Safety Agency (EASA); or Airbus's EASA Design Organization Approval (DOA). If approved by the DOA, the approval must include the DOA-authorized signature.

(m) Related Information

(1) Refer to Mandatory Continuing Airworthiness Information (MCAI) EASA Airworthiness Directive 2014–0194, dated October 15, 2014, for related information. This MCAI may be found in the AD docket

on the Internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA–2015–8427.

(2) Service information identified in this AD that is not incorporated by reference is available at the addresses specified in paragraphs (n)(6) and (n)(7) of this AD.

(n) Material Incorporated by Reference

(1) The Director of the Federal Register approved the incorporation by reference (IBR) of the service information listed in this paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) You must use this service information as applicable to do the actions required by this AD, unless this AD specifies otherwise.

(3) The following service information was approved for IBR on June 10, 2016.

(i) Airbus A300–600 Airworthiness Limitations Section (ALS), Part 5—Fuel Airworthiness Limitations, Revision 00, dated May 27, 2014. The issue date of this document is not identified on the title page.

(ii) Reserved.

(4) The following service information was approved for IBR on December 28, 2009 (74 FR 65398, December 10, 2009).

(i) Airbus A300–600 Fuel Airworthiness Limitations, Document 95A.1929/05, Issue 2, dated May 16, 2007.

(ii) Reserved.

(5) The following service information was approved for IBR on June 27, 2007 (72 FR 28827, May 23, 2007).

(i) Airbus A300–600 ALS Part 5—Fuel Airworthiness Limitations, dated May 31, 2006.

(ii) Airbus A300–600 Fuel Airworthiness Limitations, Issue 1, dated December 19, 2005.

(6) For service information identified in this AD, contact Airbus SAS, Airworthiness Office—EAW, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France; telephone +33 5 61 93 36 96; fax +33 5 61 93 44 51; email account.airworth-eas@airbus.com; Internet <http://www.airbus.com>.

(7) You may view this service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, WA. For information on the availability of this material at the FAA, call 425–227–1221.

(8) You may view this service information that is incorporated by reference at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call 202–741–6030, or go to: <http://www.archives.gov/federal-register/cfr/ibr-locations.html>.

Issued in Renton, Washington, on April 21, 2016.

Michael Kaszycki,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 2016–10159 Filed 5–5–16; 8:45 am]

BILLING CODE 4910–13–P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 71

[Docket No. FAA–2016–4235; Airspace Docket No. 16–ASW–6]

Amendment of Class E Airspace; Ash Flat, AR

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final rule.

SUMMARY: This action amends the airspace designation and airport name in Class E airspace by changing the designation from Cherokee Village Airport, AR, to Ash Flat, AR; and changing the airport name from Cherokee Village Airport to Sharp County Village Airport, Ash Flat, AR. These changes reflect the current information in the FAA's aeronautical database.

DATES: Effective 0901 UTC, July 21, 2016. The Director of the Federal Register approves this incorporation by reference action under Title 1, Code of Federal Regulations, part 51, subject to the annual revision of FAA Order 7400.9 and publication of conforming amendments.

ADDRESSES: FAA Order 7400.9Z, Airspace Designations and Reporting Points, and subsequent amendments can be viewed online at http://www.faa.gov/air_traffic/publications/. For further information, you can contact the Airspace Policy Group, Federal Aviation Administration, 800 Independence Avenue SW., Washington, DC 20591; telephone: 202–267–8783. The Order is also available for inspection at the National Archives and Records Administration (NARA). For information on the availability of FAA Order 7400.9Z at NARA, call 202–741–6030, or go to http://www.archives.gov/federal_register/code_of_federal_regulations/ibr_locations.html.

FAA Order 7400.9, Airspace Designations and Reporting Points, is published yearly and effective on September 15.

FOR FURTHER INFORMATION CONTACT: Jeffrey Claypool, Federal Aviation Administration, Operations Support Group, Central Service Center, 10101 Hillwood Parkway, Fort Worth, TX, 76177; telephone (817) 222–5711.

SUPPLEMENTARY INFORMATION:

Authority for This Rulemaking

The FAA's authority to issue rules regarding aviation safety is found in Title 49 of the United States Code.