

and is likely to exist or develop on other products of the same type design. This proposed AD would require removal of affected fuel feed pumps.

### Costs of Compliance

We estimate that this proposed AD affects 190 engines installed on airplanes of U.S. registry. We also estimate that it would take about 0.5 hours per engine to comply with this proposed AD. The average labor rate is \$85 per hour. Pro-rated cost of life limit reduction would be about \$160 per part. Based on these figures, we estimate the cost of the proposed AD on U.S. operators to be \$38,475.

### Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. "Subtitle VII: Aviation Programs," describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in "Subtitle VII, Part A, Subpart III, Section 44701: General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

### Regulatory Findings

We determined that this proposed AD would not have federalism implications under Executive Order 13132. This proposed AD would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify this proposed regulation:

(1) Is not a "significant regulatory action" under Executive Order 12866,

(2) Is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979),

(3) Will not affect intrastate aviation in Alaska to the extent that it justifies making a regulatory distinction, and

(4) Will not have a significant economic impact, positive or negative, on a substantial number of small entities

under the criteria of the Regulatory Flexibility Act.

### List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

### The Proposed Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA proposes to amend 14 CFR part 39 as follows:

### PART 39—AIRWORTHINESS DIRECTIVES

- 1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

#### § 39.13 [Amended]

- 2. The FAA amends § 39.13 by adding the following new airworthiness directive (AD):

**Technify Motors GmbH (Type Certificate previously held by Thielert Aircraft Engines GmbH):** Docket No. FAA-2015-5193; Directorate Identifier 2015-NE-35-AD.

#### (a) Comments Due Date

We must receive comments by March 4, 2016.

#### (b) Affected ADs

None.

#### (c) Applicability

This AD applies to all Technify Motors GmbH (type certificate previously held by Thielert Aircraft Engines GmbH) TAE 125-02-99 and TAE 125-02-114 reciprocating engines with a fuel feed pump, part number (P/N) 05-7312-K0073xx, or P/N 05-7312-K0133xx, where "xx" can be any number, installed.

#### (d) Reason

This AD was prompted by reports of in-flight shutdowns (IFSDs) on TAE 125-02 engines. We are issuing this AD to prevent failure of the fuel feed pump, which could result in damage to the engine and damage to the airplane.

#### (e) Actions and Compliance

Comply with this AD within the compliance times specified, unless already done. Remove from service each affected fuel feed pump before it exceeds 600 operating hours (OH) time in service (TIS) or within 110 OH after the effective date of this AD, whichever occurs later.

#### (f) Installation Prohibition

After the effective date of this AD, do not install onto any engine, any fuel feed pump, P/N 05-7312-K0073xx or P/N 05-7312-K0133xx, where "xx" can be any number, if the fuel feed pump has 600 hours or more TIS. If TIS of a fuel feed pump is unknown or has exceeded 600 hours TIS, then the fuel feed pump is not eligible for installation.

Rebuilt, overhauled, or repaired fuel feed pumps and/or fuel feed pumps that lack a serial number, are not eligible for installation.

### (g) Alternative Methods of Compliance (AMOCs)

The Manager, Engine Certification Office, FAA, may approve AMOCs for this AD. Use the procedures found in 14 CFR 39.19 to make your request. You may email your request to: [ANE-AD-AMOC@faa.gov](mailto:ANE-AD-AMOC@faa.gov).

### (h) Related Information

(1) For more information about this AD, contact Philip Haberlen, Aerospace Engineer, Engine Certification Office, FAA, Engine & Propeller Directorate, 12 New England Executive Park, Burlington, MA 01803; phone: 781-238-7770; fax: 781-238-7199; email: [philip.haberlen@faa.gov](mailto:philip.haberlen@faa.gov).

(2) Refer to MCAI European Aviation Safety Agency AD 2015-0189, dated September 21, 2015, for more information. You may examine the MCAI in the AD docket on the Internet at <http://www.regulations.gov> by searching for and locating it in Docket No. FAA-2015-5193.

(3) For service information identified in this proposed AD, contact Technify Motors GmbH, Platanenstrasse 14, D-09356 Sankt Egidien, Germany; phone: +49-37204-696-0; fax: +49-37204-696-2912; email: [support@continentaldiesel.de](mailto:support@continentaldiesel.de).

(4) You may view this service information at the FAA, Engine & Propeller Directorate, 12 New England Executive Park, Burlington, MA. For information on the availability of this material at the FAA, call 781-238-7125.

Issued in Burlington, Massachusetts, on December 18, 2015.

**Ann C. Mollica,**

*Acting Directorate Manager, Engine & Propeller Directorate, Aircraft Certification Service.*

[FR Doc. 2015-32962 Filed 12-31-15; 8:45 am]

**BILLING CODE 4910-13-P**

## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 39

[Docket No. FAA-2015-7532; Directorate Identifier 2015-NM-069-AD]

RIN 2120-AA64

### Airworthiness Directives; Dassault Aviation

**AGENCY:** Federal Aviation Administration (FAA), DOT.

**ACTION:** Notice of proposed rulemaking (NPRM).

**SUMMARY:** We propose to adopt a new airworthiness directive (AD) for all Dassault Aviation Model FALCON 7X airplanes. This proposed AD was prompted by reports of multiple cases of ram air turbine (RAT) blade damage.

This proposed AD would require deployment of the RAT, replacement of the RAT placard with a new RAT placard, and re-identification of the RAT. We are proposing this AD to prevent blade damage to the RAT which could prevent RAT deployment in flight during an emergency, possibly resulting in reduced control of the airplane.

**DATES:** We must receive comments on this proposed AD by February 18, 2016.

**ADDRESSES:** You may send comments, using the procedures found in 14 CFR 11.43 and 11.45, by any of the following methods:

- *Federal eRulemaking Portal:* Go to <http://www.regulations.gov>. Follow the instructions for submitting comments.
- *Fax:* 202-493-2251.
- *Mail:* U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE., Washington, DC 20590.
- *Hand Delivery:* U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE., Washington, DC, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

For service information identified in this proposed AD, contact Dassault Falcon Jet, P.O. Box 2000, South Hackensack, NJ 07606; telephone: 201-440-6700; Internet <http://www.dassaultfalcon.com>. You may view this referenced service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, WA. For information on the availability of this material at the FAA, call 425-227-1221.

#### Examining the AD Docket

You may examine the AD docket on the Internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA-2015-7532; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this proposed AD, the regulatory evaluation, any comments received, and other information. The street address for the Docket Operations office (telephone: 800-647-5527) is in the **ADDRESSES** section. Comments will be available in the AD docket shortly after receipt.

**FOR FURTHER INFORMATION CONTACT:** Tom Rodriguez, Aerospace Engineer, International Branch, ANM-116, Transport Airplane Directorate, FAA, 1601 Lind Avenue SW., Renton, WA 98057-3356; telephone: 425-227-1137; fax: 425-227-1149.

#### SUPPLEMENTARY INFORMATION:

##### Comments Invited

We invite you to send any written relevant data, views, or arguments about this proposed AD. Send your comments to an address listed under the **ADDRESSES** section. Include “Docket No. FAA-2015-7532; Directorate Identifier 2015-NM-069-AD” at the beginning of your comments. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of this proposed AD. We will consider all comments received by the closing date and may amend this proposed AD based on those comments.

We will post all comments we receive, without change, to <http://www.regulations.gov>, including any personal information you provide. We will also post a report summarizing each substantive verbal contact we receive about this proposed AD.

##### Discussion

The European Aviation Safety Agency (EASA), which is the Technical Agent for the Member States of the European Union, has issued EASA AD 2015-0076, dated May 6, 2015 (referred to after this as the Mandatory Continuing Airworthiness Information, or “the MCAI”), to correct an unsafe condition for all Dassault Aviation Model FALCON 7X airplanes. The MCAI states:

A few cases of Ram Air Turbine (RAT) blade damage have been reported during maintenance operations. This kind of damage is caused by an incorrect locking of RAT rotor, due to improper positioning of blades at beginning of retraction, and locking check during retraction, which likely occurs during stowage of the RAT, after its deployment for maintenance purposes.

This condition, if not corrected, could prevent RAT deployment in flight during an emergency, possibly resulting in reduced control of the aeroplane.

To address this potential unsafe condition, Dassault Aviation issued Service Bulletin (SB) 7X-289, which provides instructions to smoothly deploy the RAT and install an improved placard to ensure proper RAT stowage/retraction after maintenance.

For the reasons described above, this [EASA] AD requires replacement of the existing RAT placard with a new placard and RAT re-identification. This [EASA] AD also provides conditions for installation of a RAT on an aeroplane.

You may examine the MCAI in the AD docket on the Internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA-2015-7532.

#### Related Service Information Under 1 CFR Part 51

Dassault Aviation has issued Dassault Mandatory Service Bulletin 7X-289, dated January 21, 2015. The service information describes procedures for deployment of the RAT, replacement of the RAT placard with a new RAT placard, and re-identification of the RAT. This service information is reasonably available because the interested parties have access to it through their normal course of business or by the means identified in the **ADDRESSES** section of this NPRM.

#### FAA's Determination and Requirements of This Proposed AD

This product has been approved by the aviation authority of another country, and is approved for operation in the United States. Pursuant to our bilateral agreement with the State of Design Authority, we have been notified of the unsafe condition described in the MCAI and service information referenced above. We are proposing this AD because we evaluated all pertinent information and determined an unsafe condition exists and is likely to exist or develop on other products of the same type design.

#### Costs of Compliance

We estimate that this proposed AD affects 45 airplanes of U.S. registry. We also estimate that it would take about 4 work-hours per product to comply with the basic requirements of this proposed AD. The average labor rate is \$85 per work-hour. Required parts would cost about \$121 per product. Based on these figures, we estimate the cost of this proposed AD on U.S. operators to be \$20,745, or \$461 per product.

According to the manufacturer, some of the costs of this proposed AD may be covered under warranty, thereby reducing the cost impact on affected individuals. We do not control warranty coverage for affected individuals. As a result, we have included all costs in our cost estimate.

#### Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. “Subtitle VII: Aviation Programs,” describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in “Subtitle VII, Part A, Subpart III, Section 44701: General requirements.” Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in

air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

### Regulatory Findings

We determined that this proposed AD would not have federalism implications under Executive Order 13132. This proposed AD would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify this proposed regulation:

1. Is not a "significant regulatory action" under Executive Order 12866;
2. Is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979);
3. Will not affect intrastate aviation in Alaska; and
4. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

### List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

### The Proposed Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA proposes to amend 14 CFR part 39 as follows:

### PART 39—AIRWORTHINESS DIRECTIVES

- 1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

#### § 39.13 [Amended]

- 2. The FAA amends § 39.13 by adding the following new airworthiness directive (AD):

**Dassault Aviation:** Docket No. FAA-2015-7532; Directorate Identifier 2015-NM-069-AD.

#### (a) Comments Due Date

We must receive comments by February 18, 2016.

#### (b) Affected ADs

None.

### (c) Applicability

This AD applies to Dassault Aviation Model FALCON 7X airplanes, certificated in any category, all serial numbers.

### (d) Subject

Air Transport Association (ATA) of America Code 24, Electrical power.

### (e) Reason

This AD was prompted by reports of multiple cases of ram air turbine (RAT) blade damage. We are issuing this AD to prevent blade damage to the RAT which could prevent RAT deployment in flight during an emergency, possibly resulting in reduced control of the airplane.

### (f) Compliance

Comply with this AD within the compliance times specified, unless already done.

### (g) Placard Replacement

Except as provided by paragraph (h) of this AD: Within 28 months or during the next accomplishment of the RAT functional test, whichever occurs first after the effective date of this AD, deploy the RAT, replace the RAT placard with a new RAT placard, and re-identify the RAT part number (P/N) 1705673A to a part number identified in paragraph (g)(1) or (g)(2) of this AD, in accordance with the Accomplishment Instructions of Dassault Mandatory Service Bulletin 7X-289, dated January 21, 2015.

(1) Change P/N 1705673A to P/N 1705673B.

(2) Change P/N 1705673A to a part number that is approved as a replacement for P/N 1705673A and approved as part of the type design by the Manager, International Branch, ANM-116, Transport Airplane Directorate, FAA; or the European Aviation Safety Agency (EASA); or Dassault Aviation's EASA Design Organization Approval (DOA); after the issue date of Dassault Mandatory Service Bulletin 7X-289, dated January 21, 2015.

### (h) Exception to Paragraph (g) of This AD

An airplane on which Dassault Aviation Modification M1428 has been embodied in production is not affected by the requirements of paragraph (g) of this AD, provided no RAT P/N 1705673A has been installed on that airplane since first flight.

### (i) Parts Installation Prohibition

As of the effective date of this AD, no person may install a RAT, part number 1705673A, on any airplane.

### (j) Other FAA AD Provisions

The following provisions also apply to this AD:

(1) *Alternative Methods of Compliance (AMOCs):* The Manager, International Branch, ANM-116, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the International Branch, send it to ATTN: Tom Rodriguez, Aerospace Engineer, International Branch, ANM-116, Transport

Airplane Directorate, FAA, 1601 Lind Avenue SW., Renton, WA 98057-3356; telephone: 425-227-1137; fax: 425-227-1149. Information may be emailed to: 9-ANM-116-AMOC-REQUESTS@faa.gov. Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office. The AMOC approval letter must specifically reference this AD.

(2) *Contacting the Manufacturer:* For any requirement in this AD to obtain corrective actions from a manufacturer, the action must be accomplished using a method approved by the Manager, International Branch, ANM-116, Transport Airplane Directorate, FAA; or the European Aviation Safety Agency (EASA); or Dassault Aviation's EASA Design Organization Approval (DOA). If approved by the DOA, the approval must include the DOA-authorized signature.

### (k) Related Information

(1) Refer to Mandatory Continuing Airworthiness Information (MCAI) EASA AD 2015-0076, dated May 6, 2015, for related information. This MCAI may be found in the AD docket on the Internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA-2015-7532.

(2) For service information identified in this AD, contact Dassault Falcon Jet, P.O. Box 2000, South Hackensack, NJ 07606; telephone: 201-440-6700; Internet <http://www.dassaultfalcon.com>. You may view this service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, WA. For information on the availability of this material at the FAA, call 425-227-1221.

Issued in Renton, Washington, on December 18, 2015.

**Jeffrey E. Duven,**

*Manager, Transport Airplane Directorate, Aircraft Certification Service.*

[FR Doc. 2015-32891 Filed 12-31-15; 8:45 am]

**BILLING CODE 4910-13-P**

## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

### 14 CFR Part 39

[Docket No. FAA-2015-5539; Directorate Identifier 2015-NE-37-AD]

RIN 2120-AA64

### Airworthiness Directives; Turbomeca S.A. Turboshift Engines

**AGENCY:** Federal Aviation Administration (FAA), DOT.

**ACTION:** Notice of proposed rulemaking (NPRM).

**SUMMARY:** We propose to adopt a new airworthiness directive (AD) for all Turbomeca S.A. Arriel 2E turboshaft engines. This proposed AD was