

Proposed Rules

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This section of the FEDERAL REGISTER contains notices to the public of the proposed issuance of rules and regulations. The purpose of these notices is to give interested persons an opportunity to participate in the rule making prior to the adoption of the final rules.

DEPARTMENT OF AGRICULTURE

Agricultural Marketing Service

7 CFR Part 205

[Doc. No. AMS–NOP–15–0012; NOP–15–06]

RIN 0581–AD44

National Organic Program (NOP); Organic Livestock and Poultry Practices Proposed Rule; Extension of Comment Period

AGENCY: Agricultural Marketing Service, USDA.

ACTION: Proposed rule; Extension of comment period.

SUMMARY: The Agricultural Marketing Service (AMS) published a proposed rule in the **Federal Register** on April 13, 2016 (71 FR 21956) on Organic Livestock and Poultry Practices. This document extends the comment period for the proposed rule for 30 days from June 13, 2016 to July 13, 2016. Multiple stakeholders requested that AMS extend the comment period due to the complexity and importance of the proposal. AMS is granting an extension.

DATES: All comments must be received on or before July 13, 2016.

ADDRESSES: Interested parties may submit written comments on the Organic Livestock and Poultry Practices proposed rule using one of the following methods:

- **Federal eRulemaking Portal:** <http://www.regulations.gov>. Follow the instructions for submitting comments.
- **Mail:** Paul Lewis Ph.D., Director Standards Division, National Organic Program, USDA–AMS–NOP, Room 2646–So., Ag Stop 0268, 1400 Independence Ave. SW., Washington, DC 20250–0268.

Instructions: All submissions received must include the docket number AMS–NOP–15–0012; NOP–15–06PR, and/or Regulatory Information Number (RIN) 0581–AD44 for this rulemaking. Commenters should identify the topic and section of the proposed rule to which their comment refers. All

commenters should refer to the *General Information* section in the Notice of Proposed Rulemaking for more information on preparing your comments. All comments received will be posted without change to <http://www.regulations.gov>.

Docket: For access to the docket, including background documents and comments received, go to <http://www.regulations.gov>. Comments submitted in response to this proposed rule will also be available for viewing in person at USDA–AMS, National Organic Program, Room 2646–South Building, 1400 Independence Ave. SW., Washington, DC, from 9 a.m. to 12 noon and from 1 p.m. to 4 p.m., Monday through Friday (except official Federal holidays). Persons wanting to visit the USDA South Building to view comments received in response to this proposed rule are requested to make an appointment in advance by calling (202) 720–3252.

FOR FURTHER INFORMATION CONTACT: Paul Lewis, Ph.D., Director, Standards Division. Telephone: (202) 720–3252; Fax: (202) 260–9151.

SUPPLEMENTARY INFORMATION: This notice extends the public comment period provided in the proposed rule published in the **Federal Register** on April 13, 2016 (71 FR 21956) on Organic Livestock and Poultry Practices. In the proposed rule, AMS solicits public comments generally and requests comments on specific topics. AMS is extending the comment period, which was set to end on June 13, 2016, to July 13, 2016.

AMS is proposing to amend the organic livestock and poultry production requirements by: Adding new provisions for livestock handling and transport for slaughter and avian living conditions; and expanding and clarifying existing requirements covering livestock health care practices and mammalian living conditions.

To submit comments, or access the proposed rule docket, please follow the instructions provided under the **ADDRESSES** section. If you have questions, consult the person listed under **FOR FURTHER INFORMATION CONTACT**.

Authority: 7 U.S.C. 6501–6522.

Dated: June 2, 2016.

Elanor Starmer,
Administrator, Agricultural Marketing Service.

[FR Doc. 2016–13537 Filed 6–7–16; 8:45 am]

BILLING CODE P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA–2016–7003; Directorate Identifier 2016–CE–015–AD]

RIN 2120–AA64

Airworthiness Directives; PILATUS AIRCRAFT LTD. Airplanes

AGENCY: Federal Aviation Administration (FAA), Department of Transportation (DOT).

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: We propose to adopt a new airworthiness directive (AD) for all PILATUS AIRCRAFT LTD. Models PC–12, PC–12/45, PC–12/47, and PC–12/47E airplanes that would supersede AD 2014–22–01. This proposed AD results from mandatory continuing airworthiness information (MCAI) originated by an aviation authority of another country to identify and correct an unsafe condition on an aviation product. The MCAI describes the unsafe condition as a need to incorporate new revisions into the Limitations section, Chapter 4, of the FAA-approved maintenance program (e.g., maintenance manual). We are issuing this proposed AD to require actions to address the unsafe condition on these products.

DATES: We must receive comments on this proposed AD by July 25, 2016.

ADDRESSES: You may send comments by any of the following methods:

- **Federal eRulemaking Portal:** Go to <http://www.regulations.gov>. Follow the instructions for submitting comments.
- **Fax:** (202) 493–2251.
- **Mail:** U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue SE., Washington, DC 20590.
- **Hand Delivery:** U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room

W12-140, 1200 New Jersey Avenue SE., Washington, DC 20590, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

For service information identified in this proposed AD, contact PILATUS AIRCRAFT LTD., Customer Service Manager, CH-6371 STANS, Switzerland; telephone: +41 (0) 41 619 33 33; fax: +41 (0) 41 619 73 11; Internet: <http://www.pilatus-aircraft.com> or email: SupportPC12@pilatus-aircraft.com You may review copies of the referenced service information at the FAA, Small Airplane Directorate, 901 Locust, Kansas City, Missouri 64106. For information on the availability of this material at the FAA, call (816) 329-4148.

Examining the AD Docket

You may examine the AD docket on the Internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA-2016-7003; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this proposed AD, the regulatory evaluation, any comments received, and other information. The street address for the Docket Office (telephone (800) 647-5527) is in the **ADDRESSES** section. Comments will be available in the AD docket shortly after receipt.

FOR FURTHER INFORMATION CONTACT: Doug Rudolph, Aerospace Engineer, FAA, Small Airplane Directorate, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone: (816) 329-4059; fax: (816) 329-4090; email: doug.rudolph@faa.gov.

SUPPLEMENTARY INFORMATION:

Comments Invited

We invite you to send any written relevant data, views, or arguments about this proposed AD. Send your comments to an address listed under the **ADDRESSES** section. Include “Docket No. FAA-2016-7003; Directorate Identifier 2016-CE-015-AD” at the beginning of your comments. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of this proposed AD. We will consider all comments received by the closing date and may amend this proposed AD because of those comments.

We will post all comments we receive, without change, to <http://www.regulations.gov>, including any personal information you provide. We will also post a report summarizing each substantive verbal contact we receive about this proposed AD.

Discussion

On October 20, 2014, we issued AD 2014-22-01, Amendment 39-18005 (79 FR 67343, November 13, 2014) (“AD 2014-22-01”). That AD required actions intended to address an unsafe condition on all PILATUS AIRCRAFT LTD. Models PC-12, PC-12/45, PC-12/47, and PC-12/47E airplanes and was based on mandatory continuing airworthiness information (MCAI) originated by an aviation authority of another country.

Since we issued AD 2014-22-01, PILATUS AIRCRAFT LTD. has issued revisions to the Limitations section of the airplane maintenance manual to include repetitive inspections of the main landing gear (MLG) attachment bolts.

The European Aviation Safety Agency (EASA), which is the Technical Agent for the Member States of the European Community, has issued AD No. 2016-0083, dated April 28, 2016 (referred to after this as “the MCAI”), to correct an unsafe condition for the specified products. The MCAI states:

The airworthiness limitations are currently defined and published in the Pilatus PC-12 Aircraft Maintenance Manual(s) (AMM) under Chapter 4, Structural, Component and Miscellaneous—Airworthiness Limitations Section (ALS) documents. The limitations contained in these documents have been identified as mandatory for continued airworthiness.

Failure to comply with these instructions could result in an unsafe condition.

EASA issued AD 2014-0170 requiring the actions as specified in ALS, Chapter 4 of AMM report 02049 issue 28, for PC-12, PC-12/45 and PC-12/47 aeroplanes, and Chapter 4 of AMM report 02300 issue 11, for PC-12/47E aeroplanes.

Since that AD was issued, Pilatus issued Chapter 4 of PC-12 AMM report 02049 issue 31, and Chapter 4 of PC-12 AMM report 02300 issue 14 (hereafter collectively referred to as ‘the applicable ALS’ in this AD), to incorporate new six-year and ten-year inspection intervals for several main landing gear (MLG) attachment bolts, and an annual inspection interval for the MLG shock absorber attachment bolts, which was previously included in the AMM Chapter 5 annual inspection. After a further review of the in-service data, Pilatus issued Service Letter (SL) 186, extending the special compliance time applicable for the MLG bolts inspection.

For the reasons described above, this AD retains the requirements of EASA AD 2014-0170, which is superseded, and requires the accomplishment of the new maintenance tasks, as described in the applicable ALS.

You may examine the MCAI on the Internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA-2016-7003.

Related Service Information Under 1 CFR Part 51

PILATUS AIRCRAFT LTD. has issued Structural, Component and Miscellaneous—Airworthiness Limitations, document 12-A-04-00-00-00A-000A-A, dated September 28, 2015, and Structural and Component Limitations—Airworthiness Limitations, document 12-B-04-00-00-00A-000A-A, dated September 28, 2015. This service information is reasonably available because the interested parties have access to it through their normal course of business or by the means identified in the **ADDRESSES** section of this NPRM.

FAA’s Determination and Requirements of the Proposed AD

This product has been approved by the aviation authority of another country, and is approved for operation in the United States. Pursuant to our bilateral agreement with this State of Design Authority, they have notified us of the unsafe condition described in the MCAI and service information referenced above. We are proposing this AD because we evaluated all information and determined the unsafe condition exists and is likely to exist or develop on other products of the same type design.

Costs of Compliance

We estimate that this AD will affect 770 products of U.S. registry. We also estimate that it would take about 1.5 work-hours per product to comply with the basic requirements of this proposed AD. The average labor rate is \$85 per work-hour. Based on these figures, we estimate the cost of this proposed AD on U.S. operators to be \$98,175, or \$127.50 per product. This breaks down as follows:

- Incorporating new revisions into the Limitations section, Chapter 4, of the FAA-approved maintenance program (e.g., maintenance manual): .5 work-hour for a fleet cost of \$32,725, or \$42.50 per product.

- New inspections: MLG attachment bolts: 1 work-hour with no parts cost for fleet cost of \$65,450 or \$85 per product.

In addition, we estimate that any necessary corrective actions (on-condition costs) that must be taken based on the proposed inspections, would take about 1 work-hour and require parts costing approximately \$100 for a cost of \$185 per product. We have no way of determining the number of products that may need these necessary corrective actions. This breaks down as follows:

The only costs that would be imposed by this proposed AD over that already

required by AD 2014–22–01 is insertion of the revised Limitation section and the MLG attachment bolts inspection and replacement as necessary.

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. "Subtitle VII: Aviation Programs," describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in "Subtitle VII, part A, subpart III, section 44701: General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

We determined that this proposed AD would not have federalism implications under Executive Order 13132. This proposed AD would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify this proposed regulation:

(1) Is not a "significant regulatory action" under Executive Order 12866,

(2) Is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979),

(3) Will not affect intrastate aviation in Alaska, and

(4) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

The Proposed Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA proposes to amend 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

■ 2. The FAA amends § 39.13 by removing Amendment 39–18005 (79 FR 67343, November 13, 2014), and adding the following new AD:

PILATUS AIRCRAFT LTD.: Docket No. FAA–2016–7003; Directorate Identifier 2016–CE–015–AD.

(a) Comments Due Date

We must receive comments by July 25, 2016.

(b) Affected ADs

This AD replaces AD 2014–22–01, 39–18005 (79 FR 67343, November 13, 2014).

(c) Applicability

This AD applies to PILATUS AIRCRAFT LTD. Models PC–12, PC–12/45, PC–12/47, and PC–12/47E airplanes, all manufacturer serial numbers (MSNs), certificated in any category.

(d) Subject

Air Transport Association of America (ATA) Code 5: Time Limits.

(e) Reason

This AD was prompted by mandatory continuing airworthiness information (MCAI) originated by an aviation authority of another country to identify and correct an unsafe condition on an aviation product. The MCAI describes the unsafe condition as a need to incorporate new revisions into the Limitations section, Chapter 4, of the FAA-approved maintenance program (e.g., maintenance manual). The limitations were revised to include repetitive inspections of the main landing gear (MLG) attachment bolts. These actions are required to ensure the continued operational safety of the affected airplanes.

(f) Actions and Compliance

Unless already done, do the actions in paragraphs (f)(1) through (f)(6) of this AD:

(1) Before further flight after July 13, 2016 (the effective date of this AD), insert the following revisions into the Limitations section of the FAA-approved maintenance program (e.g., maintenance manual):

(i) STRUCTURAL, COMPONENT AND MISCELLANEOUS—AIRWORTHINESS LIMITATIONS, Data module code 12–A–04–00–00–00A–000A–A, dated September 28, 2015, of the Pilatus Model type—PC–12, PC–12/45, PC–12/47, Aircraft Maintenance Manual (AMM), Document No. 02049, 12–A–AM–00–00–00–I, revision 31, dated November 10, 2015; and

(ii) STRUCTURAL AND COMPONENT LIMITATIONS—AIRWORTHINESS LIMITATIONS, Data module code 12–B–04–00–00–00A–000A–A, dated September 28, 2015, of the Pilatus Model type—PC–12/47E MSN–1001–UP, Aircraft Maintenance

Manual (AMM), Document No. 02300, 12–B–AM–00–00–00–I, revision 14, dated November 6, 2015.

(2) The limitations section revisions listed in paragraphs (f)(1)(i) and (f)(1)(ii) of this AD do the following:

(i) Establish inspections of the MLG attachment bolts,

(ii) Specify replacement of components before or upon reaching the applicable life limit, and

(iii) Specify accomplishment of all applicable maintenance tasks within certain thresholds and intervals.

(3) Only authorized Pilatus Service Centers can do the Supplemental Structural Inspection Document (SSID) as required by the documents in paragraphs (f)(1)(i) and (f)(1)(ii) of this AD because deviations from the type design in critical locations could make the airplane ineligible for this life extension.

(4) If no compliance time is specified in the documents listed in paragraphs (f)(1)(i) and (f)(1)(ii) of this AD when doing any corrective actions where discrepancies are found as required in paragraph (f)(2)(iii) of this AD, do these corrective actions before further flight after doing the applicable maintenance task.

(5) During the accomplishment of the actions required in paragraph (f)(2) of this AD, including all subparagraphs, if a discrepancy is found that is not identified in the documents listed in paragraphs (f)(1)(i) and (f)(1)(ii) of this AD, before further flight after finding the discrepancy, contact PILATUS AIRCRAFT LTD. at the address specified in paragraph (h) of this AD for a repair scheme and incorporate that repair scheme.

(6) Within the next 6 years time-in-service after July 13, 2016 (the effective date of this AD) or within the next 3 months TIS after July 13, 2016 (the effective date of this AD), whichever occurs later, inspect the MLG attachment bolts for cracks and corrosion take all necessary corrective actions.

(g) Other FAA AD Provisions

The following provisions also apply to this AD:

(1) *Alternative Methods of Compliance (AMOCs):* The Manager, Standards Office, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. Send information to ATTN: Doug Rudolph, Aerospace Engineer, FAA, Small Airplane Directorate, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone: (816) 329–4059; fax: (816) 329–4090; email: doug.rudolph@faa.gov.

(i) Before using any approved AMOC on any airplane to which the AMOC applies, notify your appropriate principal inspector (PI) in the FAA Flight Standards District Office (FSDO), or lacking a PI, your local FSDO.

(ii) AMOCs approved for AD 2014–22–01, 39–18005 (79 FR 67343, November 13, 2014) are not approved as AMOCs for this AD.

(2) *Airworthy Product:* For any requirement in this AD to obtain corrective actions from a manufacturer or other source, use these actions if they are FAA-approved. Corrective actions are considered FAA-approved if they are approved by the State of Design Authority

(or their delegated agent). You are required to assure the product is airworthy before it is returned to service.

(h) Related Information

Refer to MCAI European Aviation Safety Agency (EASA) AD No. 2016-0083, dated April 28, 2016, for related information. You may examine the MCAI on the Internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA-2016-7003. For service information related to this AD, contact PILATUS AIRCRAFT LTD., Customer Service Manager, CH-6371 STANS, Switzerland; telephone: +41 (0) 41 619 33 33; fax: +41 (0) 41 619 73 11; Internet: <http://www.pilatus-aircraft.com> or email: SupportPC12@pilatus-aircraft.com. You may review copies of the referenced service information at the FAA, Small Airplane Directorate, 901 Locust, Kansas City, Missouri 64106. For information on the availability of this material at the FAA, call (816) 329-4148.

Issued in Kansas City, Missouri, on May 27, 2016.

Melvin Johnson,

Acting Manager, Small Airplane Directorate, Aircraft Certification Service.

[FR Doc. 2016-13249 Filed 6-7-16; 8:45 am]

BILLING CODE 4910-13-P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA-2016-6897; Directorate Identifier 2015-NM-187-AD]

RIN 2120-AA64

Airworthiness Directives; Bombardier, Inc. Airplanes

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: We propose to supersede Airworthiness Directive (AD) 2015-03-01, for all Bombardier, Inc. Model CL-600-2B19 (Regional Jet Series 100 & 440) airplanes. AD 2015-03-01 currently requires installing additional attaching hardware on the left and right fan cowl access panels and the nacelle attaching structures. Since we issued AD 2015-03-01, there have been updates to the weight and balance data needed to calculate the center of gravity for affected airplanes. This proposed AD would require weight and balance data to be included in the Weight and Balance Manual and applicable logbooks for airplanes modified per Bombardier Service Bulletin 601R-71-034, Revision B, dated August 1, 2014. The proposed AD would also require the weight and balance data to be used

in order to calculate the center of gravity for affected airplanes. We are proposing this AD to prevent damage to the fuselage and flight control surfaces from dislodged engine fan cowl panels, and prevent incorrect weight and balance calculations. Incorrect weight and balance calculations may shift the center of gravity beyond approved design parameters and affect in-flight control, which could endanger passengers and crew.

DATES: We must receive comments on this proposed AD by July 25, 2016.

ADDRESSES: You may send comments by any of the following methods:

- **Federal eRulemaking Portal:** Go to <http://www.regulations.gov>. Follow the instructions for submitting comments.
- **Fax:** 202-493-2251.
- **Mail:** U.S. Department of

Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE., Washington, DC 20590.

- **Hand Delivery:** U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE., Washington, DC, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

For service information identified in this NPRM, contact Bombardier, Inc., 400 Côte Vertu Road West, Dorval, Québec H4S 1Y9, Canada; telephone 514-855-5000; fax 514-855-7401; email thd.crj@aero.bombardier.com; Internet <http://www.bombardier.com>. You may view this referenced service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, WA. For information on the availability of this material at the FAA, call 425-227-1221.

Examining the AD Docket

You may examine the AD docket on the Internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA-2016-6897; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this proposed AD, the regulatory evaluation, any comments received, and other information. The street address for the Docket Operations office (telephone 800-647-5527) is in the **ADDRESSES** section. Comments will be available in the AD docket shortly after receipt.

FOR FURTHER INFORMATION CONTACT: Aziz Ahmed, Aerospace Engineer, Airframe and Mechanical Systems Branch, ANE-171, FAA, New York Aircraft Certification Office (ACO), 1600 Stewart

Avenue, Suite 410, Westbury, NY 11590; telephone 516-228-7329; fax 516-794-5531.

SUPPLEMENTARY INFORMATION:

Comments Invited

We invite you to send any written relevant data, views, or arguments about this proposed AD. Send your comments to an address listed under the **ADDRESSES** section. Include "Docket No. FAA-2016-6897; Directorate Identifier 2015-NM-187-AD" at the beginning of your comments. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of this proposed AD. We will consider all comments received by the closing date and may amend this proposed AD based on those comments.

We will post all comments we receive, without change, to <http://www.regulations.gov>, including any personal information you provide. We will also post a report summarizing each substantive verbal contact we receive about this proposed AD.

Discussion

On January 30, 2015, we issued AD 2015-03-01, Amendment 39-18097 (80 FR 7298, February 10, 2015) ("AD 2015-03-01"). AD 2015-03-01 requires installing additional attaching hardware on the left and right fan cowl access panels and the nacelle attaching structures. Since we issued AD 2015-03-01, we have determined it is necessary to update the weight and balance data needed to calculate the center of gravity for affected airplanes.

Transport Canada Civil Aviation (TCCA), which is the aviation authority for Canada, has issued Canadian Airworthiness Directive CF-2014-20R1, dated August 12, 2015 (referred to after this as the Mandatory Continuing Airworthiness Information, or "the MCAI"), to correct an unsafe condition for all Bombardier, Inc. Model CL-600-2B19 (Regional Jet Series 100 & 440) airplanes. The MCAI states:

There have been a number of engine fan cowl panel dislodgement incidents reported on the Bombardier CL-600-2B19 aeroplane fleet. The dislodged panels may cause damage to the fuselage and flight control surfaces of the aeroplane. Also, the debris from a dislodged panel may result in runway contamination and has the potential of causing injury on the ground.

Although the majority of the subject panel dislodgements were reported on the first or second flight after an engine maintenance task was performed that required removal and installation of the subject panels, the frequency of the dislodgements indicates that the existing attachment design is prone to human (maintenance) error.