To facilitate participation in the listening sessions, individuals wishing to present oral comments should call the designated contact person for each location to schedule an appearance. Dependent upon the number of individuals wishing to participate, oral comments may be limited. All information presented at the listening sessions will be recorded and included in the public record of the comments on the proposed rule.

This notice is issued pursuant to the provisions of the Agricultural Marketing Agreement Act of 1937, as amended (7 U.S.C. 601–674), and the applicable rules of practice and procedure governing the formulation of marketing agreements and marketing orders (7 CFR Part 900).

Dated: March 10, 1998.

#### Enrique F. Figueroa,

Administrator, Agricultural Marketing Service.

[FR Doc. 98–6583 Filed 3–12–98; 8:45 am] BILLING CODE 3410–02–M

#### DEPARTMENT OF TRANSPORTATION

#### **Federal Aviation Administration**

#### 14 CFR Part 39

[Docket No. 97-CE-146-AD]

RIN 2120-AA64

## Airworthiness Directives; Aermacchi, S.p.A. S205 Series and Models S208 and S208A Airplanes

**AGENCY:** Federal Aviation Administration, DOT.

**ACTION:** Notice of proposed rulemaking

(NPRM).

**SUMMARY:** This document proposes to adopt a new airworthiness directive (AD) that would apply to certain Aermacchi, S.p.A. (Aermacchi) S205 series and Models S208 and S208A airplanes. The proposed action would require inspecting the flaps cable pulley bracket for the correct alignment and correcting if mis-aligned; inspecting the flaps control cable for wear (nicks, cuts, frays, etc.), and replacing the flaps control pulley bracket and flap control cable if worn. The proposed AD is the result of mandatory continuing airworthiness information (MCAI) issued by the airworthiness authority for Italy. The actions specified by the proposed AD are intended to prevent flap control failure which, if not corrected, could result in loss of control of the airplane.

**DATES:** Comments must be received on or before April 24, 1998.

ADDRESSES: Submit comments in triplicate to the Federal Aviation Administration (FAA), Central Region, Office of the Regional Counsel, Attention: Rules Docket No. 97–CE–146–AD, Room 1558, 601 E. 12th Street, Kansas City, Missouri 64106. Comments may be inspected at this location between 8 a.m. and 4 p.m., Monday through Friday, holidays excepted.

Service information that applies to the proposed AD may be obtained from SIAI Marchetti, Product Support, Via Indipendenza 2, 21018 Sesto Calende (VA), Italy: telephone: +39–331–929117; facsimile: +39–331–922525. This information also may be examined at the Rules Docket at the address above.

FOR FURTHER INFORMATION CONTACT: Mr. David O. Keenan, Project Officer, FAA, Small Airplane Directorate, Aircraft Certification Service, 1201 Walnut, suite 900, Kansas City, Missouri 64106; telephone (816) 426–6934; facsimile (816) 426–2169.

#### SUPPLEMENTARY INFORMATION:

#### **Comments Invited**

Interested persons are invited to participate in the making of the proposed rule by submitting such written data, views, or arguments as they may desire. Communications should identify the Rules Docket number and be submitted in triplicate to the address specified above. All communications received on or before the closing date for comments, specified above, will be considered before taking action on the proposed rule. The proposals contained in this notice may be changed in light of the comments received.

Comments are specifically invited on the overall regulatory, economic, environmental, and energy aspects of the proposed rule. All comments submitted will be available, both before and after the closing date for comments, in the Rules Docket for examination by interested persons. A report that summarizes each FAA-public contact concerned with the substance of this proposal will be filed in the Rules Docket.

Commenters wishing the FAA to acknowledge receipt of their comments submitted in response to this notice must submit a self-addressed, stamped postcard on which the following statement is made: "Comments to Docket No. 97–CE–14–AD." The postcard will be date stamped and returned to the commenter.

#### **Availability of NPRMs**

Any person may obtain a copy of this NPRM by submitting a request to the

FAA, Central Region, Office of the Regional Counsel, Attention: Rules Docket No. 97–CE–146–AD, Room 1558, 601 E. 12th Street, Kansas City, Missouri 64106

#### **Discussion**

The Registro Aeronautico Italiano (R.A.I.), which is the airworthiness authority for Italy, recently notified the FAA that an unsafe condition may exist on certain Aermacchi S205 series and Models S208 and S208A airplanes. The R.A.I. reports an incident of flap control failure during flight on one of these Aermacchi airplanes. The investigation revealed that the flaps control pulley bracket and flaps control cables were worn. Signs of wear on the pulley bracket would be defined as misalignment of the bracket. Wear on the flaps control cable would be defined as cuts, frays, nicks, etc. These conditions, if not corrected, could result in flap control failure, which, could cause loss of control of the airplane.

#### **Relevant Service Information**

SIAI Marchetti has issued Service Bulletin No. 205B60, dated July 24, 1995, which specifies procedures for: inspecting the flaps control pulley bracket for alignment, and correcting any mis-alignment; inspecting the flaps control pulley cable for wear, and replacing these parts if worn.

The R.A.I. classified this service bulletin as mandatory and issued Italian AD 95–237, dated August 29, 1995, in order to assure the continued airworthiness of these airplanes in Italy.

## The FAA's Determination

These airplane models are manufactured in Italy and are type certificated for operation in the United States under the provisions of section 21.29 of the Federal Aviation Regulations (14 CFR 21.29) and the applicable bilateral airworthiness agreement. Pursuant to this bilateral airworthiness agreement, the R.A.I. has kept the FAA informed of the situation described above.

The FAA has examined the findings of the R.A.I.; reviewed all available information, including the service information referenced above; and determined that AD action is necessary for products of this type design that are certificated for operation in the United States

# **Explanation of the Provisions of the Proposed AD**

Since an unsafe condition has been identified that is likely to exist or develop in other Aermacchi S205 series and Models S208 and S208A airplanes

of the same type design registered in the United States, the proposed AD would require: inspecting the flaps control pulley bracket for alignment; correcting any mis-alignment; inspecting the flaps control pulley cable for wear; and, replacing the bracket and cable if worn. Accomplishment of the proposed action would be in accordance with SIAI Marchetti Service Bulletin No. 205B60, dated July 24, 1995.

#### **Cost Impact**

The FAA estimates that 70 airplanes in the U.S. registry would be affected by the proposed AD, that it would take approximately 4 workhours per airplane to accomplish the proposed action, and that the average labor rate is approximately \$60 an hour. Parts cost approximately \$150 per airplane. Based on these figures, the total cost impact of the proposed AD on U.S. operators is estimated to be \$27,300 or \$390 per airplane.

### Regulatory Impact

The regulations proposed herein would not have substantial direct effects on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, in accordance with Executive Order 12612, it is determined that this proposal would not have sufficient federalism implications to warrant the preparation of a Federalism Assessment.

For the reasons discussed above, I certify that this action (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) if promulgated, will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A copy of the draft regulatory evaluation prepared for this action has been placed in the Rules Docket. A copy of it may be obtained by contacting the Rules Docket at the location provided under the caption ADDRESSES.

#### **List of Subjects in 14 CFR Part 39**

Air transportation, Aircraft, Aviation safety, Safety.

## **The Proposed Amendment**

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration proposes to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

## PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

### § 39.13 [Amended]

2. Section 39.13 is amended by adding a new airworthiness directive (AD) to read as follows:

Aermacchi, S.P.A.: Docket No. 97-CE-146-AD.

Applicability: Models S205–18/F, S205–18/R, S205–20/F, S205–20/R, S205–22/R, S208, and S208A airplanes (all serial numbers), certificated in any category.

Note 1: This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (d) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

*Compliance*: Required within the next 100 hours time-in-service (TIS) after the effective date of this AD, unless already accomplished.

To prevent flap control failure which, if not corrected, could result in loss of control of the airplane, accomplish the following:

(a) Inspect the flaps cable pulley bracket for mis-alignment, and if misaligned, prior to further flight, replace the pulley bracket in accordance with the Instructions section of SIAI Marchetti Service Bulletin No. 205B60, dated July 24, 1995.

(b) Inspect the flaps control cable for wear (cuts, nicks, frays, etc.), and if wear is found, prior to further flight, replace the control cable in accordance with the Instructions section of SIAI Marchetti Service Bulletin No. 205B60, dated July 24, 1995.

(c) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

(d) An alternative method of compliance or adjustment of the compliance time that provides an equivalent level of safety may be approved by the Manager, Small Airplane Directorate, 1201 Walnut, suite 900, Kansas City, Missouri 64106. The request shall be forwarded through an appropriate FAA Maintenance Inspector, who may add comments and then send it to the Manager, Small Airplane Directorate.

**Note 2:** Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Small Airplane Directorate.

(e) Questions or technical information related to SIAI Marchetti Mandatory Service

Bulletin No. 205B60, dated July 24, 1995, should be directed to SIAI Marchetti, Product Support, Via Indipendenza 2, 21018 Sesto Calende (VA), Italy: telephone: +39–331–929117; facsimile: +39–331–922525. This service information may be examined at the FAA, Central Region, Office of the Regional Counsel, Room 1558, 601 E. 12th Street, Kansas City, Missouri 64106.

**Note 3:** The subject of this AD is addressed in Italian AD 95–237, dated August 29, 1995.

Issued in Kansas City, Missouri, on March 5, 1998.

#### James E. Jackson,

Acting Manager, Small Airplane Directorate, Aircraft Certification Service.

[FR Doc. 98–6451 Filed 3–12–98; 8:45 am] BILLING CODE 4910–13–U

### **DEPARTMENT OF TRANSPORTATION**

#### **Federal Aviation Administration**

#### 14 CFR Part 39

[Docket No. 97-SW-49-AD]

Airworthiness Directives; Eurocopter France Model SA-365N1, AS-365N2, and SA-366G1 Helicopters

**AGENCY:** Federal Aviation Administration, DOT.

**ACTION:** Notice of proposed rulemaking (NPRM).

**SUMMARY:** This document proposes the adoption of a new airworthiness directive (AD) that is applicable to Eurocopter France (Eurocopter) Model SA-365N1, AS-365N2, and SA-366G1 helicopters. This proposal would require initial and repetitive inspections of the tail rotor blade Kevlar tie-bar (Kevlar tie-bar) for cracks or delaminations. This proposal is prompted by a report of delamination of a Kevlar tie-bar. The actions specified by the proposed AD are intended to detect cracks that could lead to delamination of the Kevlar tie-bar, loss of tail rotor control, and subsequent loss of control of the helicopter.

**DATES:** Comments must be received by April 13, 1998.

ADDRESSES: Submit comments in triplicate to the Federal Aviation Administration (FAA), Office of the Regional Counsel, Southwest Region, Attention: Rules Docket No. 97–SW–49–AD, 2601 Meacham Blvd, Room 663, Fort Worth, Texas 76197. Comments may be inspected at this location between 9:00 a.m. and 3:00 p.m., Monday through Friday, except Federal holidays.

This service information referenced in the proposed rule may be obtained from American Eurocopter Corporation, 2701 Forum Drive, Grand Prairie, Texas