

Regulatory Impact

The regulations adopted herein will not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, it is determined that this final rule does not have federalism implications under Executive Order 13132.

For the reasons discussed above, I certify that this action (1) Is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A final evaluation has been prepared for this action and it is contained in the Rules Docket. A copy of it may be obtained from the Rules Docket at the location provided under the caption "ADDRESSES."

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:

2000-07-08 Boeing: Amendment 39-11662. Docket 99-NM-232-AD.

Applicability: Model 777 series airplanes, line numbers 2 through 103 inclusive, 105 through 119 inclusive, 121 through 161 inclusive, 163 through 177 inclusive, and 179 through 186 inclusive; certificated in any category.

Note 1: This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an

alternative method of compliance in accordance with paragraph (b) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Required as indicated, unless accomplished previously.

To prevent broken tie rods, which could result in the center stowage bins dropping onto the passenger seats below, causing possible injury to the occupants, accomplish the following:

Replacement

(a) Within 4 years after the effective date of this AD, replace the aluminum clevis ends on the tie rods for the center stowage bin supports with new steel clevis ends, in accordance with the Accomplishment Instructions of Boeing Service Bulletin 777-25-0120, Revision 1, dated March 16, 2000.

Note 2: Accomplishment of the replacement of clevis ends with new steel clevis ends prior to the effective date of this AD in accordance with Boeing Service Bulletin 777-25-0120, dated February 11, 1999, is acceptable for compliance with paragraph (a) of this AD.

Alternative Methods of Compliance

(b) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, Seattle Aircraft Certification Office (ACO), FAA, Transport Airplane Directorate. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, Seattle ACO.

Note 3: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Seattle ACO.

Special Flight Permits

(c) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

Incorporation by Reference

(d) The replacement shall be done in accordance with Boeing Service Bulletin 777-25-0120, Revision 1, dated March 16, 2000. This incorporation by reference was approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Copies may be obtained from Boeing Commercial Airplane Group, P.O. Box 3707, Seattle, Washington 98124-2207. Copies may be inspected at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

Effective Date

(e) This amendment becomes effective on May 16, 2000.

Issued in Renton, Washington, on March 31, 2000.

Donald L. Riggins,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 00-8513 Filed 4-10-00; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 99-NM-205-AD; Amendment 39-11661; AD 2000-07-07]

RIN 2120-AA64

Airworthiness Directives; Airbus Model A300 Series Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule.

SUMMARY: This amendment adopts a new airworthiness directive (AD), applicable to certain Airbus Model A300 series airplanes, that requires modification of wing center box angle fittings at frame 47. This amendment is prompted by issuance of mandatory continuing airworthiness information by a foreign civil airworthiness authority. The actions specified by this AD are intended to prevent reduced structural integrity of the wing center box angle fittings at frame 47 due to fatigue cracking.

DATES: Effective May 16, 2000.

The incorporation by reference of certain publications listed in the regulations is approved by the Director of the Federal Register as of May 16, 2000.

ADDRESSES: The service information referenced in this AD may be obtained from Airbus Industrie, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France. This information may be examined at the Federal Aviation Administration (FAA), Transport Airplane Directorate, Rules Docket, 1601 Lind Avenue, SW., Renton, Washington; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

FOR FURTHER INFORMATION CONTACT: Norman B. Martenson, Manager, International Branch, ANM-116, FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 227-2110; fax (425) 227-1149.

SUPPLEMENTARY INFORMATION: A proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) to

include an airworthiness directive (AD) that is applicable to certain Airbus Model A300 series airplanes was published in the **Federal Register** on January 27, 2000 (65 FR 4386). That action proposed to require modification of wing center box angle fittings at frame 47.

Comments

Interested persons have been afforded an opportunity to participate in the making of this amendment. Due consideration has been given to the single comment received.

The commenter states that it is not affected by the proposal.

Conclusion

After careful review of the available data, including the comment noted above, the FAA has determined that air safety and the public interest require the adoption of the rule as proposed.

Cost Impact

The FAA estimates that 38 airplanes of U.S. registry will be affected by this AD, that it will take approximately 430 work hours per airplane to accomplish the required modification, and that the average labor rate is \$60 per work hour. Required parts will cost approximately \$8,840 per airplane. Based on these figures, the cost impact of the AD on U.S. operators is estimated to be \$1,316,320, or \$34,640 per airplane.

The cost impact figure discussed above is based on assumptions that no operator has yet accomplished any of the requirements of this AD action, and that no operator would accomplish those actions in the future if this AD were not adopted.

Regulatory Impact

The regulations adopted herein will not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, it is determined that this final rule does not have federalism implications under Executive Order 13132.

For the reasons discussed above, I certify that this action (1) Is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) will not have a significant economic

impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A final evaluation has been prepared for this action and it is contained in the Rules Docket. A copy of it may be obtained from the Rules Docket at the location provided under the caption **ADDRESSES**.

List of Subjects 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:

2000-07-07 Airbus Industrie: Amendment 39-11661. Docket 99-NM-205-AD.

Applicability: Model A300 series airplanes, as listed in Airbus Service Bulletin A300-53-0298, Revision 03, dated November 26, 1998; certificated in any category; except those on which Airbus Service Bulletin A300-53-0282 or Airbus Service Bulletin A300-53-0291 has been accomplished.

Note 1: This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (c) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Required as indicated, unless accomplished previously.

To prevent reduced structural integrity of the wing center box angle fittings at frame (FR) 47, accomplish the following:

(a) Prior to the accumulation of the applicable threshold specified in the

"MANDATORY TH" column of the table in paragraph 1.B.(4) of the service bulletin, or within 6,500 flight cycles after the effective date of this AD, whichever occurs later: Except as required by paragraph (b) of this AD, modify the wing center box angle fittings at FR 47 (including removing certain sealant and fasteners, performing rotating probe inspections to detect cracking, cold working certain fastener holes, installing new fasteners and sealant, and repairing damage), in accordance with Airbus Service Bulletin A300-53-0298, Revision 03, dated November 26, 1998.

Note 2: Operators should note that the area required to be modified by paragraph (a) of this AD remains subject to the requirements of AD 96-13-11, amendment 39-9679, after modification.

(b) Where Airbus Service Bulletin A300-53-0298, Revision 03, dated November 26, 1998, specifies that Airbus be contacted for repair instructions for certain damage conditions, this AD requires that such damage conditions be repaired prior to further flight in accordance with a method approved by either the Manager, International Branch, ANM-116, FAA, Transport Airplane Directorate; or the Direction Generale de l'Aviation Civile (DGAC) (or its delegated agent). For a repair method to be approved by the Manager, International Branch, ANM-116, as required by this paragraph, the Manager's approval letter must specifically reference this AD.

Alternative Methods of Compliance

(c) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, International Branch, ANM-116, FAA, Transport Airplane Directorate. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, International Branch, ANM-116.

Note 3: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the International Branch, ANM-116.

Special Flight Permits

(d) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

Incorporation by Reference

(e) Except as provided by paragraph (b) of this AD, the modification shall be done in accordance with Airbus Service Bulletin A300-53-0298, Revision 03, dated November 26, 1998, which contains the following list of effective pages:

Page number	Revision level shown on page	Date shown on page
1-21, 32-40, 42-46, 67, 68, 71-74, 93, 94, 103-110, 151, 157-161, 205-214.	03	November 26, 1998.
22-31, 41, 47-55, 57-66, 69, 70, 75-92, 95-102, 152-156, 163-204, 215.	Original	October 14, 1993.
56, 102A, 102B, 111-150	1	March 17, 1994.

This incorporation by reference was approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Copies may be obtained from Airbus Industrie, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France. Copies may be inspected at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the Office of the Federal Register, 800 North Capitol Street, NW., Suite 700, Washington, DC.

Note 4: The subject of this AD is addressed in French airworthiness directive 1999-076-267(B), dated February 24, 1999.

(f) This amendment becomes effective on May 16, 2000.

Issued in Renton, Washington, on March 31, 2000.

Donald L. Riggins,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 00-8514 Filed 4-10-00; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 99-NM-53-AD; Amendment 39-11666; AD 2000-07-12]

RIN 2120-AA64

Airworthiness Directives; Boeing Model 727 Series Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule.

SUMMARY: This amendment adopts a new airworthiness directive (AD), applicable to certain Boeing Model 727 series airplanes, that requires repetitive structural inspections of certain aging airplanes, and repair, if necessary. This amendment also provides for optional terminating action for the repetitive inspections. This amendment is prompted by reports of incidents involving fatigue cracking and corrosion in transport category airplanes that are approaching or have exceeded their economic design service goal. The actions specified by this AD are intended to prevent degradation of the structural capabilities of the affected

airplanes. This AD relates to the recommendations of the Airworthiness Assurance Task Force assigned to review Model 727 series airplanes, which indicate that, to assure long term continued operational safety, various structural inspections should be accomplished.

DATES: Effective May 16, 2000.

The incorporation by reference of certain publications listed in the regulations is approved by the Director of the Federal Register as of May 16, 2000.

ADDRESSES: The service information referenced in this AD may be obtained from Boeing Commercial Airplane Group, P.O. Box 3707, Seattle, Washington 98124-2207. This information may be examined at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington. This information may be examined at the Federal Aviation Administration (FAA), Transport Airplane Directorate, Rules Docket, 1601 Lind Avenue, SW., Renton, Washington; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

FOR FURTHER INFORMATION CONTACT:

Walter Sippel, Aerospace Engineer, Airframe Branch, ANM-120S, FAA, Transport Airplane Directorate, Seattle Aircraft Certification Office, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 227-2774; fax (425) 227-1181.

SUPPLEMENTARY INFORMATION:

A proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) to include an airworthiness directive (AD) that is applicable to certain Boeing Model 727 series airplanes was published in the **Federal Register** on June 25, 1999 (64 FR 34168). That action proposed to require repetitive structural inspections of certain aging airplanes, and repair, if necessary.

Comments

Interested persons have been afforded an opportunity to participate in the making of this amendment. Due consideration has been given to the comments received.

Support for the Proposal

The Air Transport Association (ATA) of America, on behalf of three of its members, indicates that these members generally support the proposal. One of those members states that it does not operate any Boeing Model 727-200 series airplanes, line numbers 1 through 1214; another member has no objections to the proposed rule; and another member has no objection to the intent of the proposed rule but proposes certain clarifications.

Requests To Correct References

Two commenters state that a number of incorrect references are cited in the proposed AD. The commenters recommend changing references from "AD 94-05-04" to "AD 90-06-09" in the "Other Relevant Rulemaking" and "Differences Between Proposed Rule and Service Bulletin" sections of the proposed AD, the applicability of the proposed AD, and paragraph (d) of the proposed AD [cited as paragraphs (g)(1) and (g)(2) in the final rule]. One of the commenters contends that Revision 3 of Boeing Service Bulletin 727-57-0127, dated August 24, 1989 (which is referenced in Boeing Document Number D6-54860), clearly references repetitive inspections at intervals of 14,000 flight cycles. However, the Boeing document only specifies an inspection in accordance with Note 2 of Revision 3 of the service bulletin, and Note 2 does not refer to the repetitive inspections. Another of the commenters contends that Revision 2 of the service bulletin, dated February 13, 1976, was cited in the Boeing document and was mandated by AD 94-07-08.

Although the "Other Relevant Rulemaking" and "Differences" sections are not included in the final rule, the FAA concurs that it is necessary to change all references from "AD 94-05-04" to "AD 90-06-09" because the proposed AD incorrectly referenced AD 94-05-04. However, with regard to the correct revision number of the service bulletin, the FAA points out that AD 94-07-08 specifies Revision 3 rather than Revision 2 of the service bulletin, and that Revision 2 of the service bulletin is relevant to AD 90-06-09. To clarify the applicability of the final rule,