

which are incurred on a continuous basis; and (3) handlers are aware of this action which was unanimously recommended by the Committee at a public meeting. Also, a 30-day comment period was provided for in the proposed rule, and no comments were received.

List of Subjects in 7 CFR Part 948

Marketing Agreements, Potatoes, Reporting and recordkeeping requirements.

For the reasons set forth in the preamble, 7 CFR part 948 is amended as follows:

PART 948—IRISH POTATOES GROWN IN COLORADO

1. The authority citation for 7 CFR part 948 continues to read as follows:

Authority: 7 U.S.C. 601–674.

2. Section 948.216 is revised to read as follows:

§ 948.216 Assessment rate.

On and after September 1, 2001, an assessment rate of \$0.0035 per hundredweight is established for Colorado Area II potatoes.

Dated: October 5, 2001.

Kenneth C. Clayton,

Associate Administrator, Agricultural Marketing Service.

[FR Doc. 01–25781 Filed 10–12–01; 8:45 am]

BILLING CODE 3410–02–P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2001–NE–22; Amendment 39–12445; AD 2001–19–05]

RIN 2120–AA64

Airworthiness Directives; Rolls-Royce plc. RB211 535 Turbofan Engines, Correction

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule; request for comments, correction.

SUMMARY: This document makes a correction to Airworthiness Directive (AD) 2001–19–05 applicable to Rolls-Royce plc (RR) models RB211–535C–37, RB211–535E4–37, RB211–535E4–B–37, and RB211–535E4–B–75 turbofan engines, with radial drive steady bearing, part number (P/N) LK76084, that was published in the **Federal Register** on September 26, 2001 (66 FR 49099). A part number referenced in items (3) and (4) of Table 1 in the

regulatory information is incorrect. This document corrects that part number. In all other respects, the original document remains the same.

EFFECTIVE DATE: October 11, 2001.

FOR FURTHER INFORMATION CONTACT:

James Lawrence, Aerospace Engineer, Engine Certification Office, FAA, Engine and Propeller Directorate, 12 New England Executive Park, Burlington, MA 01803–5299; telephone (781) 238–7176, fax (781) 238–7199.

SUPPLEMENTARY INFORMATION: A final rule; request for comments airworthiness directive applicable to (RR) models RB211–535C–37, RB211–535E4–37, RB211–535E4–B–37, and RB211–535E4–B–75 turbofan engines, with radial drive steady bearing, part number (P/N) LK76084, was published in the **Federal Register** on September 26, 2001 (66 FR 49099). The following correction is needed:

PART 39—[CORRECTED]

§ 39.13 [Corrected]

On page 49100, in the third column in the Regulatory Information, Table 1, “(3) One engine has a radial drive steady bearing P/N FK76084 with fewer than 600 HSN, and the other engine has a bearing P/N FK76084 with more than 1,500 HSN,” is corrected to read “(3) One engine has a radial drive steady bearing P/N LK76084 with fewer than 600 HSN, and the other engine has a bearing P/N LK76084 with more than 1,500 HSN.” Also, Table 1 “(4) Both engines have a radial drive steady bearing P/N FK76084 with 600 or more HSN,” is corrected to read “(4) Both engines have a radial drive steady bearing P/N LK76084 with 600 or more HSN.”

Issued in Burlington, MA, on October 1, 2001.

Jay J. Pardee,

Manager, Engine and Propeller Directorate, Aircraft Certification Service.

[FR Doc. 01–25053 Filed 10–12–01; 8:45 am]

BILLING CODE 4910–13–P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2000–SW–67–AD; Amendment 39–12466; AD 2001–20–18]

RIN 2120–AA64

Airworthiness Directives; Robinson Helicopter Company Model R44 Helicopters

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule.

SUMMARY: This amendment adopts a new airworthiness directive (AD) for Robinson Helicopter Company (RHC) Model R44 helicopters that requires establishing a life limit of 2200 hours time-in-service (TIS) for affected horizontal stabilizers. This amendment is prompted by engineering analysis which indicates that certain vertical-to-horizontal stabilizer attach channels (channels) will crack sooner than the original life limit of the horizontal stabilizer. The actions specified by this AD are intended to prevent a crack through a channel, separation of the stabilizers, and subsequent loss of directional control of the helicopter.

EFFECTIVE DATE: November 19, 2001.

FOR FURTHER INFORMATION CONTACT: Fred Guerin, Aviation Safety Engineer, FAA, Los Angeles Aircraft Certification Office, Airframe Branch, 3960 Paramount Blvd., Lakewood, California 90712, telephone (562) 627–5232, fax (562) 627–5210.

SUPPLEMENTARY INFORMATION: A proposal to amend 14 CFR part 39 to include an AD for RHC R44 helicopters was published in the **Federal Register** on June 25, 2001 (66 FR 33653). That action proposed a life limit of 2200 hours TIS for affected horizontal stabilizers and removing, inspecting, and replacing certain channels with airworthy channels.

Interested persons have been afforded an opportunity to participate in the making of this amendment. No comments were received on the proposal or the FAA's determination of the cost to the public. The FAA has determined that air safety and the public interest require the adoption of the rule as proposed.

The FAA estimates that 3 helicopters of U.S. registry will be affected by this AD, that it will take approximately ½ work hour per helicopter to inspect the horizontal stabilizer and replace the channels. The average labor rate is \$60 per work hour. The manufacturer states

in their service bulletin that there will be no charge for the parts. Based on these figures, the total cost impact of the AD on U.S. operators is estimated to be \$90.

The regulations adopted herein will not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, it is determined that this final rule does not have federalism implications under Executive Order 13132.

For the reasons discussed above, I certify that this action (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A final evaluation has been prepared for this action and it is contained in the Rules Docket. A copy of it may be obtained from the Rules Docket at the FAA, Office of the Regional Counsel, Southwest Region, 2601 Meacham Blvd., Room 663, Fort Worth, Texas.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

Adoption of the Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by adding a new airworthiness directive to read as follows:

2001-20-18 Robinson Helicopter Company:
Amendment 39-12466. Docket No. 2000-SW-67-AD.

Applicability: Model R44 helicopters, with horizontal stabilizer assembly (assembly), part number (P/N) C044-1; horizontal stabilizer serial number (S/N) 0009 through 0224, except S/N 0018, 0090, 0094, 0111, 0129, 0144, 0161, 0178, 0201, and 0223, installed, certificated in any category.

Note 1: This AD applies to each helicopter identified in the preceding applicability

provision, regardless of whether it has been otherwise modified, altered, or repaired in the area subject to the requirements of this AD. For helicopters that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (c) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Required as indicated, unless accomplished previously.

To prevent a crack through a vertical-to-horizontal stabilizer attach channel (channel), which can cause separation of the stabilizers and subsequent loss of control of the helicopter, accomplish the following:

(a) Before accumulating 2200 hours time-in-service (TIS) on the assembly:

(1) Remove the vertical stabilizer to inspect the nutplate on channels, P/N D283-1 and -2.

(2) If the nutplates are P/N MS21086L4, no further action is required by this AD.

(3) If the nutplates are P/N NAS697A4, replace the channels with airworthy channels, P/N D296-1 or -2.

Note 2: Robinson Helicopter Company Service Bulletin SB-39, dated September 12, 2000, pertains to the subject of this AD.

(b) This AD revises the Limitations section of the maintenance manual by establishing a retirement life of 2200 hours TIS for assembly, P/N CO44-1, with channels, P/N D283-1 or -2, with nutplates, P/N NAS697A4, installed.

(c) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, Los Angeles Aircraft Certification Office (LAACO), FAA. Operators shall submit their requests through an FAA Principal Maintenance Inspector, who may concur or comment and then send it to the Manager, LAACO.

Note 3: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the LAACO.

(d) Special flight permits may be issued in accordance with 14 CFR 21.197 and 21.199 to operate the helicopter to a location where the requirements of this AD can be accomplished.

(e) This amendment becomes effective on November 19, 2001.

Issued in Fort Worth, Texas, on October 3, 2001.

Mark R. Schilling,

*Acting Manager, Rotorcraft Directorate,
Aircraft Certification Service.*

[FR Doc. 01-25693 Filed 10-12-01; 8:45 am]

BILLING CODE 4910-13-P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2001-SW-49-AD; Amendment 39-12470; AD 2001-19-52]

RIN 2120-AA64

Airworthiness Directives; Bell Helicopter Textron Canada Model 222, 222B, 222U, and 230 Helicopters

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule; request for comments.

SUMMARY: This document publishes in the **Federal Register** an amendment adopting Airworthiness Directive (AD) 2001-19-52 which was sent previously to all known U.S. owners and operators of Bell Helicopter Textron Canada (BHTC) Model 222, 222B, 222U, and 230 helicopters by individual letters. This AD requires removing certain serial-numbered main rotor pendulum weight supports from service and replacing with airworthy main rotor pendulum weight supports. This AD is prompted by the failure of a main rotor pendulum weight support (support) resulting in shedding of the weights and an increased level of main rotor vibration. The actions specified by this AD are intended to prevent failure of a support, loss of a weight set resulting in main rotor vibration, and subsequent loss of control of the helicopter.

DATES: Effective October 30, 2001, to all persons except those persons to whom it was made immediately effective by Emergency AD 2001-19-52, issued on September 21, 2001, which contained the requirements of this amendment.

Comments for inclusion in the Rules Docket must be received on or before December 14, 2001.

ADDRESSES: Submit comments in triplicate to the Federal Aviation Administration (FAA), Office of the Regional Counsel, Southwest Region, Attention: Rules Docket No. 2001-SW-49-AD, 2601 Meacham Blvd., Room 663, Fort Worth, Texas 76137. You may also send comments electronically to the Rules Docket at the following address: 9-asw-adcomments@faa.gov.

FOR FURTHER INFORMATION CONTACT: Jim Grigg, Aviation Safety Engineer, FAA, Rotorcraft Directorate, Regulations Group, Fort Worth, Texas 76193-0110, telephone (817) 222-5490, fax (817) 222-5961.

SUPPLEMENTARY INFORMATION: On September 21, 2001, the FAA issued Emergency AD 2001-19-52, for BHTC