

What Is the Unsafe Condition Presented in This AD?

(d) The actions specified by this AD are intended to prevent the loading of the baro

set potentiometer, which could result in an unacceptable altitude error. This condition could cause the pilot to make flight decisions that put the airplane in unsafe flight conditions.

What Must I Do To Address This Problem?

(e) To address this problem, you must do the following:

Actions	Compliance	Procedures
<p>(1) Inspect the TAWS8000 TAWS (part number 805-18000-001 that incorporates hardware "Mod None", "Mod A", or "Mod B") installation to determine if both the TAWS8000 TAWS and any other device are connected to the same baro set potentiometer.</p> <p>(2) If both the TAWS8000 TAWS and any other device are connected to the same baro set potentiometer, remove the TAWS8000 TAWS and cap and stow the connecting wires or replace the TAWS8000 TAWS unit with a unit that incorporates hardware "Mod C".</p> <p>(3) Do not install or reconfigure any TAWS8000 TAWS (part number 805-18000-001) that does not incorporate hardware "Mod C".</p>	<p>Within the next 5 hours time-in-service (TIS) after July 21, 2003 (the effective date of AD 2003-13-08), unless already done.</p> <p>Before further flight after the inspection required in paragraph (d)(1) of this AD.</p> <p>As of June 7, 2004 (the effective date of this AD).</p>	<p>Follow Goodrich Avionics Systems, Inc. Service Memo SM #134, dated May 2, 2003, or Goodrich Avionics Systems, Inc. Service Memo SM #134, revised July 9, 2003, and the applicable installation manual.</p> <p>For removing the TAWS8000 TAWS, follow Goodrich Avionics Systems, Inc. Service Memo SM #134, dated May 2, 2003, or Goodrich Avionics Systems, Inc. Service Memo SM #134, revised July 9, 2003, and the applicable installation manual. For replacing the TAWS8000 TAWS, follow Goodrich Avionics Systems, Inc. Alert Service Bulletin SB #A117, dated July 9, 2003.</p> <p>Not Applicable.</p>

May I Request an Alternative Method of Compliance?

(f) You may request a different method of compliance or a different compliance time for this AD by following the procedures in 14 CFR 39.19.

(1) Unless FAA authorizes otherwise, send your request to your principal inspector. The principal inspector may add comments and will send your request to the Manager, Chicago Aircraft Certification Office (ACO), FAA. For information on any already approved alternative methods of compliance, contact Brenda S. Ocker, Aerospace Engineer, FAA, Chicago Aircraft Certification Office, 2300 East Devon Avenue, Des Plaines, Illinois 60018; telephone: (847) 294-7126; facsimile: (847) 294-7834.

(2) Alternative methods of compliance approved under AD 2003-13-08, which is superseded by this AD, are approved as alternative methods of compliance with this AD.

Does This AD Incorporate Any Material by Reference?

(g) You must do the actions required by this AD following the instructions in Goodrich Avionics Systems, Inc. Service Memo SM #134, dated May 2, 2003; Goodrich Avionics Systems, Inc. Service Memo SM #134, revised July 9, 2003; and Goodrich Avionics Systems, Inc. Alert Service Bulletin SB #A117, dated July 9, 2003.

(1) On July 21, 2003 (68 FR 38586, June 30, 2003), and in accordance with 5 U.S.C. 552(a) and 1 CFR part 51, the Director of the Federal Register approved the incorporation by reference of Goodrich Avionics Systems, Inc. Service Memo SM #134, dated May 2, 2003.

(2) As of June 7, 2004, and in accordance with 5 U.S.C. 552(a) and 1 CFR part 51, the Director of the Federal Register approved the incorporation by reference of Goodrich Avionics Systems, Inc. Service Memo SM #134, revised July 9, 2003; and Goodrich

Avionics Systems, Inc. Alert Service Bulletin SB #A117, dated July 9, 2003.

(3) You may get a copy from Goodrich Avionics Systems, Inc., 5353 52nd Street, SE., Grand Rapids, Michigan 49512-9704; telephone: (616) 949-6600; facsimile: (616) 977-6898. You may review copies at FAA, Central Region, Office of the Regional Counsel, 901 Locust, Room 506, Kansas City, Missouri 64106; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

Issued in Kansas City, Missouri, on April 13, 2004.

James E. Jackson,

Acting Manager, Small Airplane Directorate, Aircraft Certification Service.

[FR Doc. 04-8792 Filed 4-20-04; 8:45 am]

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DEPARTMENT OF TRANSPORTATION**Federal Aviation Administration****14 CFR Part 39**

[Docket No. 2003-CE-62-AD; Amendment 39-13583; AD 2004-08-14]

RIN 2120-AA64

Airworthiness Directives; Glasflugel Models Mosquito and Club Libelle 205 Sailplanes

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final rule.

SUMMARY: The FAA adopts a new airworthiness directive (AD) for all Glasflugel Models Mosquito and Club Libelle 205 sailplanes. This AD requires you to replace the rudder actuator arm

with an improved design rudder actuator arm. This AD is the result of mandatory continuing airworthiness information (MCAI) issued by the airworthiness authority for Germany. We are issuing this AD to prevent the rudder attachment actuator arm from failing due to ground handling damage. This failure could eventually result in reduced or loss of sailplane control.

DATES: This AD becomes effective on May 28, 2004.

As of May 28, 2004, the Director of the Federal Register approved the incorporation by reference of certain publications listed in the regulation.

ADDRESSES: You may get the service information identified in this AD from Glasflugel, Glasfaser-Flugzeug-Service GmbH, Hansjory Steifeneder, Hofener Weg, 72582 Grabenstetten, Germany; telephone: 011 49 7382 1032.

You may view the AD docket at FAA, Central Region, Office of the Regional Counsel, Attention: Rules Docket No. 2003-CE-62-AD, 901 Locust, Room 506, Kansas City, Missouri 64106. Office hours are 8 a.m. to 4 p.m., Monday through Friday, except Federal holidays.

FOR FURTHER INFORMATION CONTACT: Greg Davison, Aerospace Engineer, FAA, Small Airplane Directorate, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone: (816) 329-4130; facsimile: (816) 329-4090.

SUPPLEMENTARY INFORMATION:**Discussion**

What events have caused this AD?
The Luftfahrt-Bundesamt (LBA), which is the airworthiness authority for Germany, recently notified FAA that an

unsafe condition may exist on all Glasflugel Models Mosquito and Club Libelle 205 sailplanes. The LBA reports incidents of rudder actuator arm failure. This failure is occurring through lifting the fuselage by the rudder.

Glasflugel has manufactured a new improved design rudder actuator arm that is less susceptible to such damage.

What is the potential impact if FAA took no action? Rudder attachment actuator arm failure could eventually result in reduced or loss of sailplane control.

Has FAA taken any action to this point? We issued a proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) to include an AD that would apply to all Glasflugel Models Mosquito and Club Libelle 205 sailplanes. This proposal was published in the **Federal Register** as a notice of proposed rulemaking (NPRM) on February 17, 2004 (69 FR 7382). The NPRM proposed to require you to

replace the rudder actuator arm with an improved design rudder actuator arm.

Comments

Was the public invited to comment? We provided the public the opportunity to participate in developing this AD. We received no comments on the proposal or on the determination of the cost to the public.

Conclusion

What is FAA's final determination on this issue? We have carefully reviewed the available data and determined that air safety and the public interest require adopting the AD as proposed except for minor editorial corrections. We have determined that these minor corrections:

- Are consistent with the intent that was proposed in the NPRM for correcting the unsafe condition; and
- Do not add any additional burden upon the public than was already proposed in the NPRM.

Changes to 14 CFR Part 39—Effect on the AD

How does the revision to 14 CFR part 39 affect this AD? On July 10, 2002, the FAA published a new version of 14 CFR part 39 (67 FR 47997, July 22, 2002), which governs the FAA's AD system. This regulation now includes material that relates to altered products, special flight permits, and alternative methods of compliance. This material previously was included in each individual AD. Since this material is included in 14 CFR part 39, we will not include it in future AD actions.

Costs of Compliance

How many sailplanes does this AD impact? We estimate that this AD affects 80 sailplanes in the U.S. registry.

What is the cost impact of this AD on owners/operators of the affected sailplanes? We estimate the following costs to do the replacement:

Labor cost	Parts cost	Total cost per sailplane	Total cost on U.S. operators
3 workhours × \$65 per hour = \$195	\$90	\$285	\$22,800

Regulatory Findings

Will this AD impact various entities? We have determined that this AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

Will this AD involve a significant rule or regulatory action? For the reasons discussed above, I certify that this AD:

1. Is not a "significant regulatory action" under Executive Order 12866;
2. Is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and
3. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared a summary of the costs to comply with this AD and placed it in the AD Docket. You may get a copy of

this summary by sending a request to us at the address listed under **ADDRESSES**. Include "AD Docket No. 2003-CE-62-AD" in your request.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

■ Accordingly, under the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

- 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

- 2. FAA amends § 39.13 by adding a new AD to read as follows:

2004-08-14 Glasflugel: Amendment 39-13583; Docket No. 2003-CE-62-AD.

When Does This AD Become Effective?

- (a) This AD becomes effective on May 28, 2004.

What Other ADs Are Affected by This Action?

- (b) None.

What Sailplanes Are Affected by This AD?

- (c) This AD affects the Models Mosquito and Club Libelle 205 sailplanes, all serial numbers, that are certificated in any category.

What Is the Unsafe Condition Presented in This AD?

- (d) This AD is the result of mandatory continuing airworthiness information (MCAI) issued by the airworthiness authority for Germany. The actions of this AD are intended to prevent the rudder attachment actuator arm from failing due to ground handling damage. This failure could eventually result in reduced or loss of sailplane control.

What Must I Do To Address This Problem?

- (e) To address this problem, you must do the following:

Actions	Compliance	Procedures
(1) Replace the rudder actuator arm (manufactured following drawing No. 203-45-10) with an improved design arm that is manufactured following drawing No. 203-45-10-2.	Within the next 25 hours time-in-service (TIS) after May 28, 2004 (the effective date of this AD), unless already done.	Follow Glasflugel Technical Note No. 205-22 and No. 206-21, dated October 14, 2002 (LBA-approved November 11, 2002); or Glasflugel Technical Note No. 303-23 and No. 304-10, dated October 14, 2002 (LBA-approved November 11, 2002), as applicable.
(2) Do not install any rudder actuator arm that is not manufactured following drawing No. 203-45-10-2.	As of May 28, 2004 (the effective date of this AD).	Not Applicable.

May I Request an Alternative Method of Compliance?

(f) You may request a different method of compliance or a different compliance time for this AD by following the procedures in 14 CFR 39.19. Unless FAA authorizes otherwise, send your request to your principal inspector. The principal inspector may add comments and will send your request to the Manager, Standards Office, Small Airplane Directorate, FAA.

For information on any already approved alternative methods of compliance, contact Greg Davison, Aerospace Engineer, FAA, Small Airplane Directorate, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone: (816) 329-4130; facsimile: (816) 329-4090.

Does This AD Incorporate Any Material by Reference?

(g) You must do the actions required by this AD following the instructions in Glasflugel Technical Note No. 205-22 and No. 206-21, dated October 14, 2002 (LBA-approved November 11, 2002); or Glasflugel Technical Note No. 303-23 and No. 304-10, dated October 14, 2002 (LBA-approved November 11, 2002), as applicable. The Director of the Federal Register approved the incorporation by reference of this service bulletin in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. You may get a copy from Glasflugel, Glasfaser-Flugzeug-Service GmbH, Hansjory Steifeneder, Hofener Weg, 72582 Grabenstetten, Germany; telephone: 011 49 7382 1032. You may review copies at FAA, Central Region, Office of the Regional Counsel, 901 Locust, Room 506, Kansas City, Missouri 64106; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

Is There Other Information That Relates to This Subject?

(h) German AD No. 2003-004 and No. 2003-005, both effective date: January 9, 2003, also address the subject of this AD.

Issued in Kansas City, Missouri, on April 13, 2004.

James E. Jackson,

Acting Manager, Small Airplane Directorate, Aircraft Certification Service.

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2001-NM-111-AD; Amendment 39-13574; AD 2004-08-05]

RIN 2120-AA64

Airworthiness Directives; Airbus Model A300 B2 Series Airplanes; A300 B4 Series Airplanes; A300 B4-600, B4-600R, F4-600R, and C4-605R Variant F (Collectively Called A300-600) Series Airplanes; and A310 Series Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule.

SUMMARY: This amendment supersedes an existing airworthiness directive (AD), applicable to certain Airbus Model A300 series airplanes, that currently requires either a one-time ultrasonic inspection, or repetitive visual inspections and eventual ultrasonic inspection, to detect cracking of the longitudinal skin splice above the mid-passenger door panels, and corrective actions if necessary. This amendment requires repetitive ultrasonic inspections to detect cracking of certain skin lap joints in additional areas of the fuselage and repair if necessary. This amendment also expands the applicability of the existing AD to include additional airplanes. The actions specified by this AD are intended to detect and correct cracking of certain skin lap joints, which could result in reduced structural integrity and decompression of the airplane. This action is intended to address the identified unsafe condition.

DATES: Effective May 26, 2004.

The incorporation by reference of certain publications, as listed in the regulations, is approved by the Director of the Federal Register as of May 26, 2004.

The incorporation by reference of a certain other publication, as listed in the regulations, was approved previously by

the Director of the Federal Register as of February 22, 2000 (65 FR 5756, February 7, 2000).

ADDRESSES: The service information referenced in this AD may be obtained from Airbus, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France. This information may be examined at the Federal Aviation Administration (FAA), Transport Airplane Directorate, Rules Docket, 1601 Lind Avenue, SW., Renton, Washington; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

FOR FURTHER INFORMATION CONTACT: Anthony Jopling, Aerospace Engineer, International Branch, ANM-116, FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 227-2190; fax (425) 227-1149.

SUPPLEMENTARY INFORMATION: A proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) by superseding AD 2000-02-39, amendment 39-11557 (65 FR 5756, February 7, 2000), which is applicable to certain Airbus Model A300 series airplanes, was published in the **Federal Register** on December 18, 2003 (68 FR 70464). The action proposed to continue to require either a one-time ultrasonic inspection, or repetitive visual inspections and eventual ultrasonic inspection, to detect cracking of the longitudinal skin splice above the mid-passenger door panels, and corrective actions if necessary. The action also proposed to require repetitive ultrasonic inspections to detect cracking of certain skin lap joints in additional areas of the fuselage and repair if necessary. In addition, the action proposed to expand the applicability of the existing AD to include additional airplanes.

Comments

Interested persons have been afforded an opportunity to participate in the making of this amendment. Due consideration has been given to the comment received.