Instructions for Continued Airworthiness by incorporating the instructions of Chapters 05-10-10, "Airworthiness Limitations," 05-10-20, "Certification Maintenance Requirements," and 05-10-30, "Critical Design Configuration Control Limitations (CDCCL)—Fuel System" of BAE Systems (Operations) Limited Jetstream Series 4100 AMM, Revision 29, dated February 15, 2008 (hereafter "the service information"). Thereafter, except as provided in paragraph (i) of this AD, no alternative replacement times or inspection intervals may be approved for any affected component. The revised Chapter 05-10-10 replaces the corresponding chapter specified in paragraph (f) of this AD.

(h) Where paragraph 2.A.(2) of the service information specifies that certain landing gear units "must be removed 31st March 2008," this AD requires compliance within 60 days after the effective date of this AD.

Alternative Methods of Compliance (AMOCs)

(i) The Manager, International Branch, ANM–116, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. Send information to ATTN: Todd Thompson, Aerospace Engineer, International Branch, ANM–116, Transport Airplane Directorate, FAA, 1601 Lind Avenue, SW., Renton, Washington 98057–3356; telephone (425) 227–1175; fax (425) 227–1149. Before using any approved AMOC on any airplane to which the AMOC applies, notify your appropriate principal inspector (PI) in the FAA Flight Standards District Office (FSDO), or lacking a PI, your local FSDO.

Related Information

(j) European Aviation Safety Agency (EASA) airworthiness directive 2008–0094, dated May 16, 2008, also addresses the subject of this AD.

Issued in Renton, Washington, on November 6, 2008.

Stephen P. Boyd,

Assistant Manager, Transport Airplane Directorate, Aircraft Certification Service. [FR Doc. E8–27161 Filed 11–14–08; 8:45 am]

BILLING CODE 4910-13-P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA-2008-1216; Directorate Identifier 2008-NM-111-AD]

RIN 2120-AA64

Airworthiness Directives; Bombardier Model CL-600-1A11 (CL-600), CL-600-2A12 (CL-601), and CL-600-2B16 (CL-601-3A, CL-601-3R, and CL-604) Airplanes

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: We propose to adopt a new airworthiness directive (AD) for the products listed above. This proposed AD results from mandatory continuing airworthiness information (MCAI) originated by an aviation authority of another country to identify and correct an unsafe condition on an aviation product. The MCAI describes the unsafe condition as:

* * * [S]everal cases of wing anti-ice piccolo duct failure reported on CL–600–2B19 (CRJ) aircraft. Although there have been no failures reported on Challenger aircraft, similar ducts are installed on the above Challenger models.

* * * * *

Cracking of the wing anti-ice piccolo ducts could result in air leakage, with an adverse effect on the anti-ice air distribution pattern and a possible unannunciated insufficient heat condition. * * *

The unsafe condition is anti-ice system air leakage with a possible adverse effect on the anti-ice air distribution pattern and anti-ice capability without annunciation to the flightcrew, and consequent reduced controllability of the airplane. The proposed AD would require actions that are intended to address the unsafe condition described in the MCAI.

DATES: We must receive comments on this proposed AD by December 17, 2008.

ADDRESSES: You may send comments by any of the following methods:

- Federal eRulemaking Portal: Go to http://www.regulations.gov. Follow the instructions for submitting comments.
 - Fax: (202) 493–2251.
- *Mail*: U.S. Department of Transportation, Docket Operations, M– 30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue, SE., Washington, DC 20590.
- Hand Delivery: U.S. Department of Transportation, Docket Operations, M— 30, West Building Ground Floor, Room W12–40, 1200 New Jersey Avenue, SE., Washington, DC, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

Examining the AD Docket

You may examine the AD docket on the Internet at http://www.regulations.gov; or in person at the Docket Operations office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this proposed AD, the regulatory evaluation, any comments received, and other information. The street address for the Docket Operations office (telephone (800) 647–5527) is in

the **ADDRESSES** section. Comments will be available in the AD docket shortly after receipt.

FOR FURTHER INFORMATION CONTACT: Dan Parrillo, Aerospace Engineer, Airframe and Propulsion Branch, ANE–171, FAA, New York Aircraft Certification Office, 1600 Stewart Avenue, Suite 410, Westbury, New York 11590; telephone (516) 228–7305; fax (516) 794–5531.

SUPPLEMENTARY INFORMATION:

Comments Invited

We invite you to send any written relevant data, views, or arguments about this proposed AD. Send your comments to an address listed under the ADDRESSES section. Include "Docket No. FAA-2008-1216; Directorate Identifier 2008-NM-111-AD" at the beginning of your comments. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of this proposed AD. We will consider all comments received by the closing date and may amend this proposed AD based on those comments.

We will post all comments we receive, without change, to http://www.regulations.gov, including any personal information you provide. We will also post a report summarizing each substantive verbal contact we receive about this proposed AD.

Discussion

Transport Canada Civil Aviation (TCCA), which is the aviation authority for Canada, has issued Canadian Airworthiness Directive CF–2008–18, dated May 9, 2008 (referred to after this as "the MCAI"), to correct an unsafe condition for the specified products. The MCAI states:

There have been several cases of wing antiice piccolo duct failure reported on CL–600–2B19 (CRJ) aircraft. Although there have been no failures reported on Challenger aircraft, similar ducts are installed on the above Challenger models [CL–600–1A11, CL–600–2A12, and CL–600–2B16].

Upon investigation, it has been determined that ducts manufactured since June 2000, and installed since 1 August 2000, are susceptible to cracking due to the process used to drill the holes in the ducts. These ducts were installed on CL-600-2B16 aircraft, serial numbers 5469 through 5635 in production, but may also have been installed as replacements on CL-600-1A11, CL-600-2A12 and other CL-600-2B16 aircraft.

Cracking of the wing anti-ice piccolo ducts could result in air leakage, with an adverse effect on the anti-ice air distribution pattern and a possible unannunciated insufficient heat condition. As a result, the airplane flight manual (AFM) instructions have been revised to provide proper annunciation of an insufficient heat condition, utilizing existing messages and indications, with instructions, to the pilot, to leave icing conditions if

sufficient heat cannot be achieved or maintained.

This directive mandates the amendment of the AFM procedures, in addition to checking the part numbers and serial numbers of the installed wing anti-ice piccolo ducts and replacing them as necessary.

The unsafe condition is anti-ice system air leakage with a possible adverse effect

on the anti-ice air distribution pattern and anti-ice capability without annunciation to the flightcrew, and consequent reduced controllability of the airplane. You may obtain further information by examining the MCAI in the AD docket.

Relevant Service Information

Bombardier has issued the service information listed in the "Temporary Revisions (TRs)" and "Service Bulletins" tables. The actions described in this service information are intended to correct the unsafe condition identified in the MCAI.

TEMPORARY REVISIONS (TRs)

Canadair TR—	To the—	Date—
600-1/19	Canadair Challenger Model CL-600-1A11 Airplane Flight Manual (AFM) Canadair Challenger Model CL-600-1A11 (Winglets) AFM Canadair Challenger Model CL-600-2A12 AFM, Product Support Publication (PSP) 601-1B-1 Canadair Challenger Model CL-600-2A12 AFM, PSP 601-1A-1 Canadair Challenger Model CL-600-2A12 AFM, PSP 601-1B Canadair Challenger Model CL-600-2B16 AFM, PSP 601A-1 Canadair Challenger Model CL-600-2A12 AFM Canadair Challenger Model CL-600-2B16 AFM, PSP 601A-1-1 Canadair Challenger Model CL-600-2B16 AFM, PSP 601A-1-1 Canadair Challenger Model CL-604 AFM, PSP 604-1	August 16, 2006. August 16, 2006. August 16, 2006. August 16, 2006. August 16, 2006. August 16, 2006.

SERVICE BULLETINS

Bombardier Service Bulletin—	Revision level—	Date—
600–0734	Original	November 30, 2006. November 30, 2006. January 21, 2008.

FAA's Determination and Requirements of This Proposed AD

This product has been approved by the aviation authority of another country, and is approved for operation in the United States. Pursuant to our bilateral agreement with the State of Design Authority, we have been notified of the unsafe condition described in the MCAI and service information referenced above. We are proposing this AD because we evaluated all pertinent information and determined an unsafe condition exists and is likely to exist or develop on other products of the same type design.

Differences Between This AD and the MCAI or Service Information

We have reviewed the MCAI and related service information and, in general, agree with their substance. But we might have found it necessary to use different words from those in the MCAI to ensure the AD is clear for U.S. operators and is enforceable. In making these changes, we do not intend to differ substantively from the information provided in the MCAI and related service information.

We might also have proposed different actions in this AD from those in the MCAI in order to follow FAA policies. Any such differences are highlighted in a Note within the proposed AD.

Costs of Compliance

Based on the service information, we estimate that this proposed AD would affect about 108 products of U.S. registry. We also estimate that it would take about 37 work-hours per product to comply with the basic requirements of this proposed AD. The average labor rate is \$80 per work-hour. Based on these figures, we estimate the cost of the proposed AD on U.S. operators to be \$319,680, or \$2,960 per product.

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. "Subtitle VII: Aviation Programs," describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in "Subtitle VII, Part A, Subpart III, Section 44701: General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on

products identified in this rulemaking action.

Regulatory Findings

We determined that this proposed AD would not have federalism implications under Executive Order 13132. This proposed AD would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify this proposed regulation:

- 1. Is not a "significant regulatory action" under Executive Order 12866;
- 2. Is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and
- 3. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared a regulatory evaluation of the estimated costs to comply with this proposed AD and placed it in the AD docket.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

The Proposed Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA proposes to amend 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. The FAA amends § 39.13 by adding the following new AD:

Bombardier, Inc. (Formerly Canadair):

Docket No. FAA-2008-1216; Directorate Identifier 2008-NM-111-AD.

Comments Due Date

(a) We must receive comments by December 17, 2008.

Affected ADs

(b) None.

Applicability

(c) This AD applies to the airplanes identified in Table 1, paragraphs (c)(1), (c)(2), (c)(3), and (c)(4) of this AD, certificated in any category.

TABLE 1—AIRPLANES AFFECTED BY THIS AD

Bombardier model	Serial Nos.
(1) CL-600-1A11 (CL-600) airplanes	1004 through 1085 inclusive. 3001 through 3066 inclusive. 5001 through 5194 inclusive. 5301 through 5635 inclusive.

Subject

(d) Air Transport Association (ATA) of America Code 30: Ice and Rain Protection.

Reason

(e) The mandatory continuing airworthiness information (MCAI) states:

There have been several cases of wing antiice piccolo duct failure reported on CL-600– 2B19 (CRJ) aircraft. Although there have been no failures reported on Challenger aircraft, similar ducts are installed on the above Challenger models.

Upon investigation, it has been determined that ducts manufactured since June 2000, and installed since 1 August 2000, are susceptible to cracking due to the process used to drill the holes in the ducts. These ducts were installed on CL-600-2B16 aircraft, serial numbers 5469 through 5635 in production, but may also have been installed as

replacements on CL-600-1A11, CL-600-2A12 and other CL-600-2B16 aircraft.

Cracking of the wing anti-ice piccolo ducts could result in air leakage, with an adverse effect on the anti-ice air distribution pattern and a possible unannunciated insufficient heat condition. As a result, the airplane flight manual (AFM) instructions have been revised to provide proper annunciation of an insufficient heat condition, utilizing existing messages and indications, with instructions, to the pilot, to leave icing conditions if sufficient heat cannot be achieved or maintained.

This directive mandates the amendment of the AFM procedures, in addition to checking the part numbers and serial numbers of the installed wing anti-ice piccolo ducts and replacing them as necessary.

The unsafe condition is anti-ice system air leakage with a possible adverse effect on the

anti-ice air distribution pattern and anti-ice capability without annunciation to the flightcrew, and consequent reduced controllability of the airplane.

Actions and Compliance

- (f) Unless already done, do the following actions.
- (1) For airplanes identified in paragraphs (c)(1), (c)(2), (c)(3), and (c)(4) of this AD: Within 30 days after the effective date of this AD, revise the Normal and Abnormal Procedures sections of the Airplane Flight Manual (AFM) by inserting a copy of the applicable Temporary Revision (TR) listed in Table 2 of this AD. When the information in the applicable TR is included in the general revisions of the AFM, the general revisions may be inserted in the AFM and the TR may be removed.

TABLE 2—TEMPORARY REVISIONS

Canadair TR—	To the—	Date—
(ii) 600–1/19	Canadair Challenger Model CL-600-1A11 Airplane Flight Manual (AFM) Canadair Challenger Model CL-600-1A11 (Winglets) AFM Canadair Challenger Model CL-600-2A12 AFM, Product Support Publication (PSP) 601-1B-1 Canadair Challenger Model CL-600-2A12 AFM, PSP 601-1A-1 Canadair Challenger Model CL-600-2A12 AFM, PSP 601-1B Canadair Challenger Model CL-600-2B16 AFM, PSP 601A-1 Canadair Challenger Model CL-600-2A12 AFM Canadair Challenger Model CL-600-2B16 AFM, PSP 601A-1-1 Canadair Challenger Model CL-600-2B16 AFM, PSP 601A-1-1 Canadair Challenger Model CL-604 AFM, PSP 604-1	August 16, 2006. August 16, 2006. April 17, 2006.

(2) For airplanes identified in paragraphs (c)(1), (c)(2), and (c)(3) of this AD, and for Model CL-600-2B16 (CL-601-3A, CL-601-3R, & CL-604) airplanes, serial numbers 5301 through 5468 inclusive: Prior to the accumulation of 2,000 total flight hours, or within 60 months after the effective date of this AD, whichever occurs first, review the airplane maintenance records to determine if any anti-ice piccolo ducts or complete leading edge sections were replaced since August 1, 2000.

(3) For airplanes identified in paragraphs (c)(1), (c)(2), and (c)(3) of this AD, and for Model CL-600-2B16 (CL-601-3A, CL-601-3R, & CL-604) airplanes, serial numbers 5301 through 5468 inclusive: If, during the action required by paragraph (f)(2) of this AD, it is determined that any anti-ice piccolo duct has been replaced since August 1, 2000, before further flight do a visual inspection to determine if any affected serial number is installed as identified in paragraph 2.C. of the applicable service bulletin identified in Table 3 of this AD. A review of airplane

maintenance records is acceptable in lieu of this inspection if the serial number of the duct can be conclusively determined from that review. If any affected serial number is installed, before further flight replace the piccolo duct with a serviceable piccolo duct that does not have a serial number identified in paragraph 2.C. of the applicable service bulletin identified in Table 3 of this AD. Do all actions in accordance with the Accomplishment Instructions of the applicable service bulletin listed in Table 3 of this AD.

TABLE	3—SERVICE	BULLETINS
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Model—	Bombardier service bul- letin—	Revision level—	Date—
(i) CL-600-1A11 (CL-600) airplanes	601–0585 601–0585	Original	November 30, 2006.

- (4) For Model CL-600-2B16 (CL-601-3A, CL-601-3R, & CL-604) airplanes, serial numbers 5469 through 5635 inclusive: Prior to the accumulation of 2,000 total flight hours, or within 60 months after the effective date of this AD, whichever occurs first, do a visual inspection of the anti-ice piccolo ducts to determine if any affected serial number identified in paragraph 2.C. of the Bombardier Service Bulletin 604-30-003, Revision 01, dated January 21, 2008 ("the service bulletin"), is installed. If any affected serial number is installed, before further flight replace the piccolo duct with a serviceable piccolo duct that does not have a serial number identified in paragraph 2.C. of the service bulletin. Do all actions in accordance with the Accomplishment Instructions of the service bulletin.
- (5) As of the effective date of this AD, no person may install on any airplane an antice piccolo duct with a serial number identified in paragraph 2.C. of the applicable service bulletin identified in Table 3 of this AD
- (6) Actions done before the effective date of this AD in accordance with Bombardier Service Bulletin 604–30–003, dated November 30, 2006, are acceptable for compliance with the corresponding actions in this AD.

FAA AD Differences

Note 1: This AD differs from the MCAI and/or service information as follows: No differences.

Other FAA AD Provisions

- (g) The following provisions also apply to this AD:
- (1) Alternative Methods of Compliance (AMOCs): The Manager, New York Aircraft Certification Office, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. Send information to ATTN: Dan Parrillo, Aerospace Engineer, Airframe and Propulsion Branch, ANE-171, FAA, New York Aircraft Certification Office, 1600 Stewart Avenue, Suite 410, Westbury, New York 11590; telephone (516) 228-7305; fax (516) 794-5531. Before using any approved AMOC on any airplane to which the AMOC applies, notify your appropriate principal inspector (PI) in the FAA Flight Standards District Office (FSDO), or lacking a PI, your local FSDO.
- (2) Airworthy Product: For any requirement in this AD to obtain corrective actions from a manufacturer or other source, use these actions if they are FAA-approved. Corrective actions are considered FAA-approved if they are approved by the State

of Design Authority (or their delegated agent). You are required to assure the product is airworthy before it is returned to service.

(3) Reporting Requirements: For any reporting requirement in this AD, under the provisions of the Paperwork Reduction Act, the Office of Management and Budget (OMB) has approved the information collection requirements and has assigned OMB Control Number 2120–0056.

Related Information

(h) Refer to MCAI Canadian Airworthiness Directive CF–2008–18, dated May 9, 2008, and the service information identified in Table 2 and Table 3 of this AD, for related information.

Issued in Renton, Washington, on November 6, 2008.

Stephen P. Boyd,

Assistant Manager, Transport Airplane Directorate, Aircraft Certification Service. [FR Doc. E8–27162 Filed 11–14–08; 8:45 am] BILLING CODE 4910–13–P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 71

[Docket No. FAA-2008-1046; Airspace Docket No. 08-ASW-21]

Proposed Amendment of Class E Airspace; Houston, TX

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Notice of proposed rulemaking.

SUMMARY: This action proposes to amend Class E airspace at Houston, TX. Additional controlled airspace is necessary to accommodate Area Navigation (RNAV) Standard Instrument Approach Procedures (SIAP) at Lone Star Executive Airport, Conroe, TX. The FAA is taking this action to enhance the safety and management of Instrument Flight Rules (IFR) aircraft operations at Lone Star Executive Airport. This action also updates the coordinates of Chambers County Airport, and reflects a name change for Scholes Field.

DATES: 0901 UTC. Comments must be

received on or before January 2, 2009. ADDRESSES: Send comments on this

ADDRESSES: Send comments on this proposal to the U.S. Department of

Transportation, Docket Operations, 1200 New Jersey Avenue, SE., West Building Ground Floor, Room W12-140, Washington, DC 20590-0001. You must identify the docket number FAA-2008-1046/Airspace Docket No. 08-ASW-21, at the beginning of your comments. You may also submit comments on the Internet at http://www.regulations.gov. You may review the public docket containing the proposal, any comments received, and any final disposition in person in the Dockets Office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The Docket Office (telephone 1-800-647-5527) is on the ground floor of the building at the above address.

FOR FURTHER INFORMATION CONTACT:

Scott Enander, Central Service Area, Operations Support Group, Federal Aviation Administration, Southwest Region, 2601 Meacham Blvd., Fort Worth, TX 76193–0530; telephone: (817) 222–5582.

SUPPLEMENTARY INFORMATION:

Comments Invited

Interested parties are invited to participate in this proposed rulemaking by submitting such written data, views, or arguments, as they may desire. Comments that provide the factual basis supporting the views and suggestions presented are particularly helpful in developing reasoned regulatory decisions on the proposal. Comments are specifically invited on the overall regulatory, aeronautical, economic, environmental, and energy-related aspects of the proposal. Communications should identify both docket numbers and be submitted in triplicate to the address listed above. Commenters wishing the FAA to acknowledge receipt of their comments on this notice must submit with those comments a self-addressed, stamped postcard on which the following statement is made: "Comments to Docket No. FAA-2008-1046/Airspace Docket No. 08-ASW-21." The postcard will be date/time stamped and returned to the commenter.