# **Proposed Rules**

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This section of the FEDERAL REGISTER contains notices to the public of the proposed issuance of rules and regulations. The purpose of these notices is to give interested persons an opportunity to participate in the rule making prior to the adoption of the final rules.

## **DEPARTMENT OF TRANSPORTATION**

## **Federal Aviation Administration**

#### 14 CFR Part 39

[Docket No. FAA-2013-0314; Directorate Identifier 2013-CE-004-AD]

#### RIN 2120-AA64

# Airworthiness Directives; B-N Group Ltd. Airplanes

**AGENCY:** Federal Aviation Administration (FAA), Department of Transportation (DOT).

**ACTION:** Notice of proposed rulemaking (NPRM).

**SUMMARY:** We propose to adopt a new airworthiness directive (AD) for B-N Group Ltd. Models BN-2, BN-2A, BN2A MK. III, BN2A MK. III-2, BN2A MK. III-3, BN-2A-2, BN-2A-20, BN-2A-21, BN-2A-26, BN-2A-27, BN-2A-3, BN-2A-6, BN-2A-8, BN-2A-9, BN-2B-20, BN-2B-21, BN-2B-26, BN-2B-27, BN-2T, and BN-2T-4R airplanes. This proposed AD results from mandatory continuing airworthiness information (MCAI) originated by an aviation authority of another country to identify and correct an unsafe condition on an aviation product. The MCAI describes the unsafe condition as inadequate sealing of the fuel filler cap (fuel tank cap) and the fuel filler receptacle (fuel tank opening), which could lead to contaminated fuel and result in in-flight shutdown of the engine. We are issuing this proposed AD to require actions to address the unsafe condition on these products.

**DATES:** We must receive comments on this proposed AD by May 24, 2013.

**ADDRESSES:** You may send comments by any of the following methods:

- Federal eRulemaking Portal: Go to http://www.regulations.gov. Follow the instructions for submitting comments.
  - Fax: (202) 493-2251.
  - Mail: U.S. Department of

Transportation, Docket Operations, M–30, West Building Ground Floor, Room

W12–140, 1200 New Jersey Avenue SE., Washington, DC 20590.

• Hand Delivery: U.S. Department of Transportation, Docket Operations, M— 30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue SE., Washington, DC 20590, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

For service information identified in this proposed AD, contact Britten-Norman Aircraft Ltd, Commodore House, Mountbatten Business Centre, Millbrook Road East, Southampton SO15 1HY, United Kingdom; telephone: +44 01983 872511; fax: +44 01983 873246; email: info@bnaircraft.com; Internet: www.britten-norman.com. You may review copies of the referenced service information at the FAA, Small Airplane Directorate, 901 Locust, Kansas City, Missouri 64106. For information on the availability of this material at the FAA, call (816) 329-4148.

## **Examining the AD Docket**

You may examine the AD docket on the Internet at http://www.regulations.gov; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this proposed AD, the regulatory evaluation, any comments received, and other information. The street address for the Docket Office (telephone (800) 647–5527) is in the ADDRESSES section. Comments will be available in the AD docket shortly after receipt.

## FOR FURTHER INFORMATION CONTACT:

Taylor Martin, Aerospace Engineer, FAA, Small Airplane Directorate, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone: (816) 329–4138; fax: (816) 329–4090; email: taylor.martin@faa.gov.

#### SUPPLEMENTARY INFORMATION:

#### **Comments Invited**

We invite you to send any written relevant data, views, or arguments about this proposed AD. Send your comments to an address listed under the

ADDRESSES section. Include "Docket No. FAA–2013–0314; Directorate Identifier 2013–CE–004–AD" at the beginning of your comments. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of this proposed AD. We will

consider all comments received by the closing date and may amend this proposed AD because of those comments.

We will post all comments we receive, without change, to <a href="http://regulations.gov">http://regulations.gov</a>, including any personal information you provide. We will also post a report summarizing each substantive verbal contact we receive about this proposed AD.

#### Discussion

The European Aviation Safety Agency (EASA), which is the Technical Agent for the Member States of the European Community, has issued AD No.: 2012–0270, December 20, 2012 (referred to after this as "the MCAI"), to correct an unsafe condition for the specified products. The MCAI states:

Preliminary investigations into a recent engine failure on a BN2 aeroplane have attributed the event to water contaminated fuel. The contamination is suspected to have occurred due to inadequate sealing between a post-mod NB–M–477 fuel filler cap and a pre-mod NB–M–477 fuel filler receptacle. This condition, if not detected and corrected, could lead to fuel water contamination, possibly resulting in in-flight shut down of the engine.

For the reasons described above, this AD requires a one-time inspection of the fuel filler cap and fuel filler receptacle to determine whether they are at the same modification state and, depending on findings, accomplishment of applicable corrective action(s). To mitigate the risk of water contamination pending the installation of matching fuel filler cap and receptacle, this AD also requires daily pre-flight water contamination checks.

You may obtain further information by examining the MCAI in the AD docket.

## **Relevant Service Information**

B–N Group Limited has issued Service Bulletin Number SB 332, Issue 1, dated December 6, 2012. The actions described in this service information are intended to correct the unsafe condition identified in the MCAI.

# FAA's Determination and Requirements of the Proposed AD

This product has been approved by the aviation authority of another country, and is approved for operation in the United States. Pursuant to our bilateral agreement with this State of Design Authority, they have notified us of the unsafe condition described in the MCAI and service information referenced above. We are proposing this AD because we evaluated all information and determined the unsafe condition exists and is likely to exist or develop on other products of the same type design.

#### **Costs of Compliance**

We estimate that this proposed AD will affect 114 products of U.S. registry. We also estimate that it would take about 1 work-hour per product to comply with the basic requirements of this proposed AD. The average labor rate is \$85 per work-hour. Required parts would cost about \$0 per product.

Based on these figures, we estimate the cost of the proposed AD on U.S. operators to be \$9,690, or \$85 per product.

In addition, we estimate that any necessary follow-on actions would take about 1 work-hour and require parts costing \$400, for a cost of \$485 per product. We have no way of determining the number of products that may need these actions.

### Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. "Subtitle VII: Aviation Programs," describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in "Subtitle VII, Part A, Subpart III, Section 44701: General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

### **Regulatory Findings**

We determined that this proposed AD would not have federalism implications under Executive Order 13132. This proposed AD would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify this proposed regulation:

(1) Is not a "significant regulatory action" under Executive Order 12866,

(2) Is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26,

(3) Will not affect intrastate aviation in Alaska, and

(4) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

#### List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safetv.

## The Proposed Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA proposes to amend 14 CFR part 39 as follows:

# PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

## § 39.13 [Amended]

■ 2. The FAA amends § 39.13 by adding the following new AD:

B-N Group Ltd.: Docket No. FAA-2013-0314; Directorate Identifier 2013-CE-004-AD.

### (a) Comments Due Date

We must receive comments by May 24, 2013.

#### (b) Affected ADs

None.

# (c) Applicability

This AD applies to B–N Group Ltd. Models BN–2, BN–2A, BN2A MK. III, BN2A MK. III–2, BN2A MK. III–3, BN–2A–2, BN–2A–20, BN–2A–21, BN–2A–26, BN–2A–27, BN–2A–3, BN–2A–6, BN–2A–8, BN–2A–9, BN–2B–20, BN–2B–21, BN–2B–26, BN–2B–27, BN–2T, and BN–2T–4R airplanes, all serial numbers, certificated in any category.

#### (d) Subject

Air Transport Association of America (ATA) Code 28: Fuel.

#### (e) Reason

This AD was prompted by mandatory continuing airworthiness information (MCAI) originated by an aviation authority of another country to identify and correct an unsafe condition on an aviation product. We are issuing this proposed AD to prevent, detect, and correct inadequate sealing of the fuel filler cap (fuel tank cap) and the fuel filler receptacle (fuel tank opening), which could lead to contaminated fuel and result in inflight shutdown of the engine.

#### (f) Actions and Compliance

Unless already done, do the following actions:

(1) Within next 30 days after the effective date of this AD, inspect the aircraft fuel replenishment points on the top surface of the wings to determine that the fuel filler cap (fuel tank cap) matches the fuel filler receptacle (fuel tank opening) following the instructions of paragraph 6 of Britten-Norman Service Bulletin Number SB 332, Issue 1, dated December 6, 2012.

(2) If a mismatch of the fuel filler cap and the fuel filler receptacle is found during the inspection required by paragraph (f)(1) of this AD, within 3 months after the effective date of this AD, install the correct fuel filler cap to match the fuel filler receptacle installed on the airplane following the instructions of paragraph 6 of Britten-Norman Service Bulletin Number SB 332, Issue 1, dated December 6, 2012.

(3) If a mismatch of the fuel filler cap and the fuel filler receptacle is found during the inspection required by paragraph (f)(1) of this AD, before further flight and thereafter during each daily pre-flight check, do water contamination checks of the gascolators and fuel tank sump drains, including those of the wing tip tanks if installed. This check is in addition to the normal daily checks already required.

(4) The modification required by paragraph (f)(2) of this AD terminates the daily preflight water contamination checks as specified in paragraph (f)(3) of this AD.

(5) After the effective date of this AD, do not install on any airplane a fuel filler cap that does not match the fuel filler receptacle and has the correct seal.

### (g) Other FAA AD Provisions

The following provisions also apply to this AD:

(1) Alternative Methods of Compliance (AMOCs): The Manager, Standards Office, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. Send information to ATTN: Taylor Martin, Aerospace Engineer, FAA, Small Airplane Directorate, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone: (816) 329–4138; fax: (816) 329–4090; email: taylor.martin@faa.gov. Before using any approved AMOC on any airplane to which the AMOC applies, notify your appropriate principal inspector (PI) in the FAA Flight Standards District Office (FSDO), or lacking a PI, your local FSDO.

(2) Airworthy Product: For any requirement in this AD to obtain corrective actions from a manufacturer or other source, use these actions if they are FAA-approved. Corrective actions are considered FAA-approved if they are approved by the State of Design Authority (or their delegated agent). You are required to assure the product is airworthy before it is returned to service.

(3) Reporting Requirements: For any reporting requirement in this AD, a federal agency may not conduct or sponsor, and a person is not required to respond to, nor shall a person be subject to a penalty for failure to comply with a collection of information subject to the requirements of the Paperwork Reduction Act unless that collection of information displays a current valid OMB Control Number. The OMB Control Number for this information

collection is 2120-0056. Public reporting for this collection of information is estimated to be approximately 5 minutes per response, including the time for reviewing instructions, completing and reviewing the collection of information. All responses to this collection of information are mandatory. Comments concerning the accuracy of this burden and suggestions for reducing the burden should be directed to the FAA at: 800 Independence Ave. SW., Washington, DC 20591, Attn: Information Collection Clearance Officer, AES-200.

#### (h) Related Information

Refer to MCAI European Aviation Safety Agency (EASA) AD No.: 2012-0270, December 20, 2012; and Britten-Norman Service Bulletin Number SB 332, Issue 1, dated December 6, 2012, for related information. For service information related to this AD, contact Britten-Norman Aircraft Ltd, Commodore House, Mountbatten Business Centre, Millbrook Road East, Southampton SO15 1HY, United Kingdom; telephone: +44 01983 872511; fax: +44 01983 873246; email: info@bnaircraft.com; Internet: www.britten-norman.com. You may review copies of the referenced service information at the FAA, Small Airplane Directorate, 901 Locust, Kansas City, Missouri 64106. For information on the availability of this material at the FAA, call (816) 329-4148.

Issued in Kansas City, Missouri, on April 2, 2013.

#### Earl Lawrence,

Manager, Small Airplane Directorate, Aircraft Certification Service.

[FR Doc. 2013-08194 Filed 4-8-13; 8:45 am]

BILLING CODE 4910-13-P

## **DEPARTMENT OF TRANSPORTATION**

# **Federal Aviation Administration**

## 14 CFR Part 39

[Docket No. FAA-2013-0296; Directorate Identifier 2012-NM-102-AD]

RIN 2120-AA64

## Airworthiness Directives; Bombardier, Inc. Airplanes

**AGENCY:** Federal Aviation Administration (FAA), DOT.

**ACTION:** Notice of proposed rulemaking

(NPRM).

**SUMMARY:** We propose to supersede an existing airworthiness directive (AD) that applies to certain Bombardier, Inc. Model CL-600-2B19 (Regional Jet Series 100 & 440) airplanes. The existing AD currently requires revising the airworthiness limitations section of the Instructions for Continued Airworthiness of the maintenance requirements manual (MRM) by incorporating procedures for repetitive functional tests of the pilot input lever

of the pitch feel simulator (PFS) units and new repetitive functional tests of the pilot input lever of the PFS unit, and corrective actions if necessary; and, after initiating the new tests, removing of the existing procedures for the repetitive functional tests from the MRM. The existing AD was prompted by a report that the shear pin located in the input lever of two PFS units failed due to fatigue. Since we issued that AD, a new re-designed PFS unit has been developed, which eliminates the need for repetitive inspections. This proposed AD would require replacing certain PFS units with the new redesigned PFS unit. This proposed AD would also remove certain airplanes from the applicability and add certain airplanes to the applicability. We are proposing this AD to prevent undetected failure of the shear pins of both PFS units simultaneously, which could result in loss of pitch feel forces and consequent reduced control of the airplane.

**DATES:** We must receive comments on this proposed AD by May 24, 2013.

**ADDRESSES:** You may send comments by any of the following methods:

- Federal eRulemaking Portal: Go to http://www.regulations.gov. Follow the instructions for submitting comments.
  - Fax: (202) 493–2251.
- Mail: U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE., Washington, DC 20590.
- Hand Delivery: U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE., Washington, DC, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

For service information identified in this proposed AD, contact Bombardier, Inc., 400 Côte-Vertu Road West, Dorval, Québec H4S 1Y9, Canada; telephone 514-855-5000; fax 514-855-7401; email thd.crj@aero.bombardier.com; Internet http://www.bombardier.com. You may review copies of the referenced service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, WA. For information on the availability of this material at the FAA, call 425-227-1221.

## **Examining the AD Docket**

You may examine the AD docket on the Internet at http:// www.regulations.gov; or in person at the Docket Operations office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this proposed AD, the regulatory evaluation, any comments

received, and other information. The street address for the Docket Operations office (telephone (800) 647-5527) is in the ADDRESSES section. Comments will be available in the AD docket shortly after receipt.

# FOR FURTHER INFORMATION CONTACT:

Cesar Gomez, Aerospace Engineer, Airframe and Mechanical Systems Branch, ANE-171, FAA, New York Aircraft Certification Office, 1600 Stewart Avenue, Suite 410, Westbury, New York 11590; telephone (516) 228-7318; fax (516) 794–5531.

#### SUPPLEMENTARY INFORMATION:

#### **Comments Invited**

We invite you to send any written relevant data, views, or arguments about this proposed AD. Send your comments to an address listed under the ADDRESSES section. Include "Docket No. FAA-2013-0296; Directorate Identifier 2012-NM-102-AD" at the beginning of your comments. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of this proposed AD. We will consider all comments received by the closing date and may amend this proposed AD based on those comments.

We will post all comments we receive, without change, to http:// www.regulations.gov, including any personal information you provide. We will also post a report summarizing each substantive verbal contact we receive about this proposed AD.

### Discussion

On March 31, 2008, we issued AD 2008-08-09, Amendment 39-15461 (73 FR 19979, April 14, 2008). That AD required actions intended to address an unsafe condition on the products listed above.

Since we issued AD 2008-08-09, Amendment 39-15461 (73 FR 19979, April 14, 2008), Transport Canada Civil Aviation (TCCA), which is the aviation authority for Canada, has issued Canadian Airworthiness Directive CF-2005-41R1, dated May 10, 2012 (referred to after this as the Mandatory Continuing Airworthiness Information, or "the MCAI"), to correct an unsafe condition for the specified products. The MCAI states:

The shear pin in the input lever of several Pitch Feel Simulators (PFS) units has failed due to fatigue. The shear pin failure is not always detectable by the flight crew in normal operation. Failure of the shear pins in both PFS units on an aeroplane could result in loss of pitch feel forces and reduced controllability of the aeroplane. Recently, Transport Canada has certified the new design of the PFS unit—part number (P/ N) 601R92300-7 as a terminating action.